

Trajectory Animator Java Applet

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I. Introduction

Low thrust trajectories allow for very complicated, programmable thrust profiles to shape the orbit. Also it is often desirable or necessary to only apply thrust at a certain point in a trajectory such as at periapsis. The Trajectory Animator allows the user to program complicated thrust profiles based on the position in the orbit (specified by true anomaly or mean anomaly), and then see how the orbit develops over time.

Trajectory Animator is a Java applet that can be run from any windows browser that supports Java. The HTML code for the applet will prompt a user to automatically download the Java plug-in if it is not already installed on the user's machine.

II. Description

A. Main Window

The main window in the Trajectory Animator consists of several parts. The top section contains the main control buttons to start/pause, define the initial orbit, define the thrust profile, define the vehicle parameters, and to restart the animation/calculations. The middle section contains zoom and speed controls as well as a button to show or hide the history of the orbit, and a button to smooth the track data (at the cost of slower performance). The bottom section contains the information pertinent to the orbit and a status label that appears when the animation reaches an end point. The main window is shown in Figure 1. The orbital information displayed include the semi-major axis (a), the eccentricity (e), the orbital energy (E), the specific angular momentum (h), the orbital period, the total elapsed time, the apoapsis (ra), the periapsis (rp), the true anomaly, the mean anomaly, the magnitude of the position vector (r), the altitude (alt), the magnitude of the velocity vector (v), the mass of the vehicle, the magnitude of the acceleration, and the flight path angle (Phi).

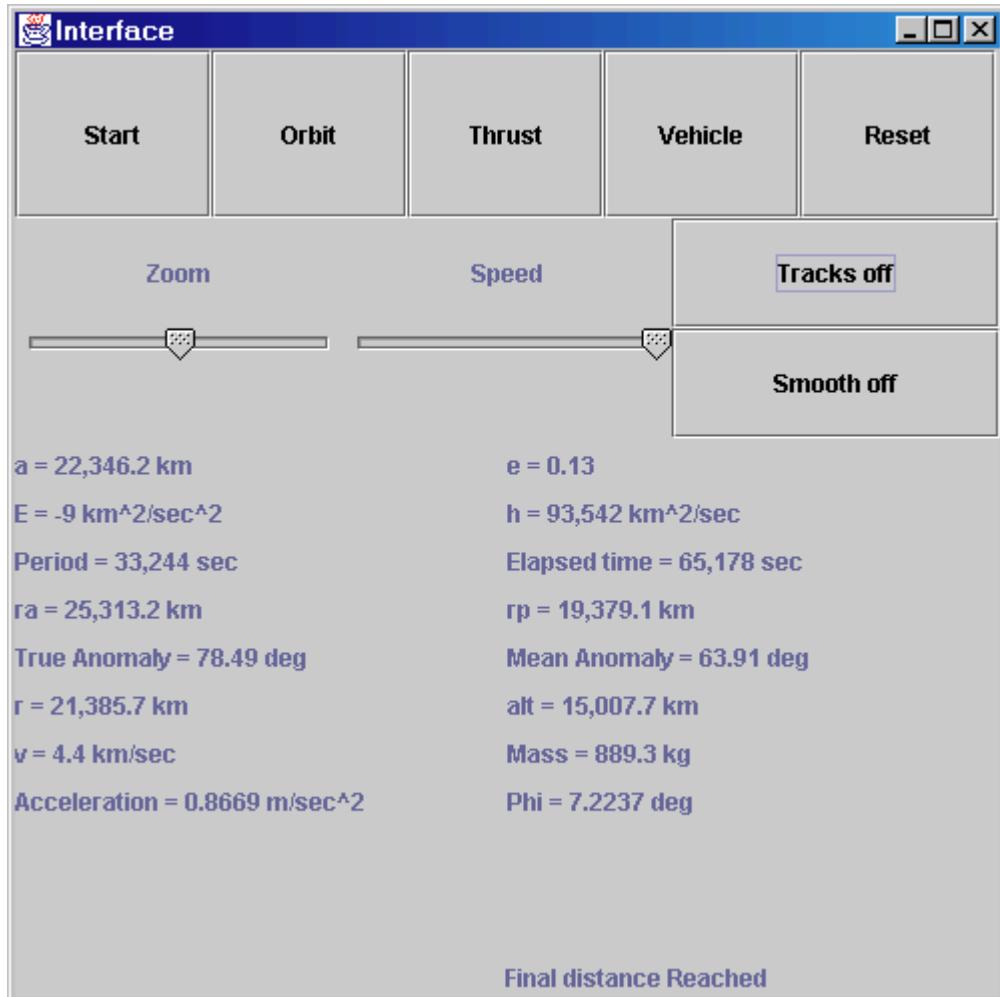


Figure 1: Main Window

B. Trajectory Animation Window

The trajectory animation window shows the sun or planet, the starting orbit in green, the final distance in red and the current position as a black circle. When the animation begins, the path that the spacecraft follows is also shown in black, unless the Tracks Off button is clicked to suppress them. A representative trajectory animation window is shown in Figure 2.

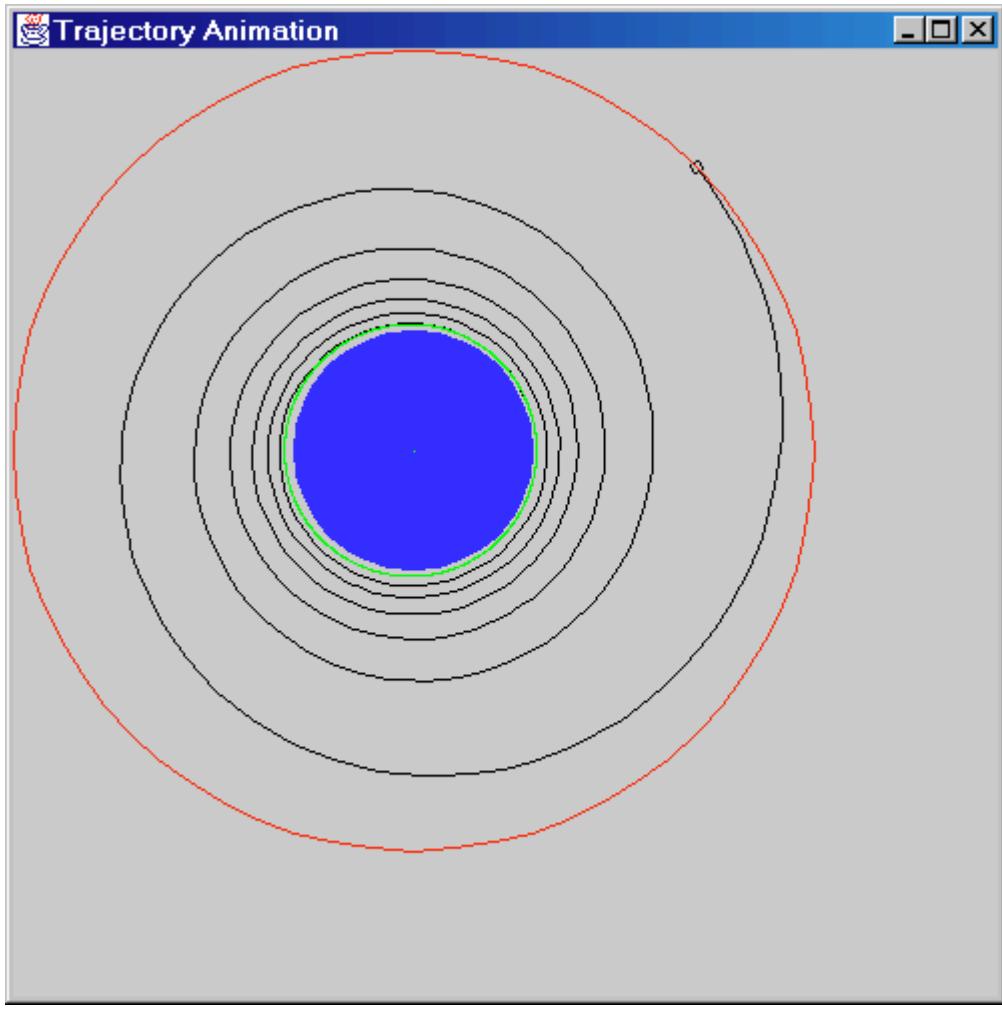


Figure 2: Trajectory Animation Window

C. Orbit Definition Window

The orbit definition window is used to define the initial orbital parameters and the final point at which the animation stops. A set of buttons allow the choice of the main body in the system. The choices are the sun and any of the planets. The unit of length can be changed from the default (meters). The initial orbit can be defined by supplying a pair of orbital parameters. The choices for the parameters are the semi-major axis and the eccentricity (a & e), the apoapsis and periapsis (ra & rp) or the apoapsis altitude and periapsis altitude (ha & hp). If the desired starting point in the orbit is not at a true anomaly of 0.0, it can be set to another value. The final position is entered as a radial

distance from the center of the planet/sun (a & e or ra & rp options) or as the final altitude (ha & hp option). The orbit definition window is shown in Figure 3.

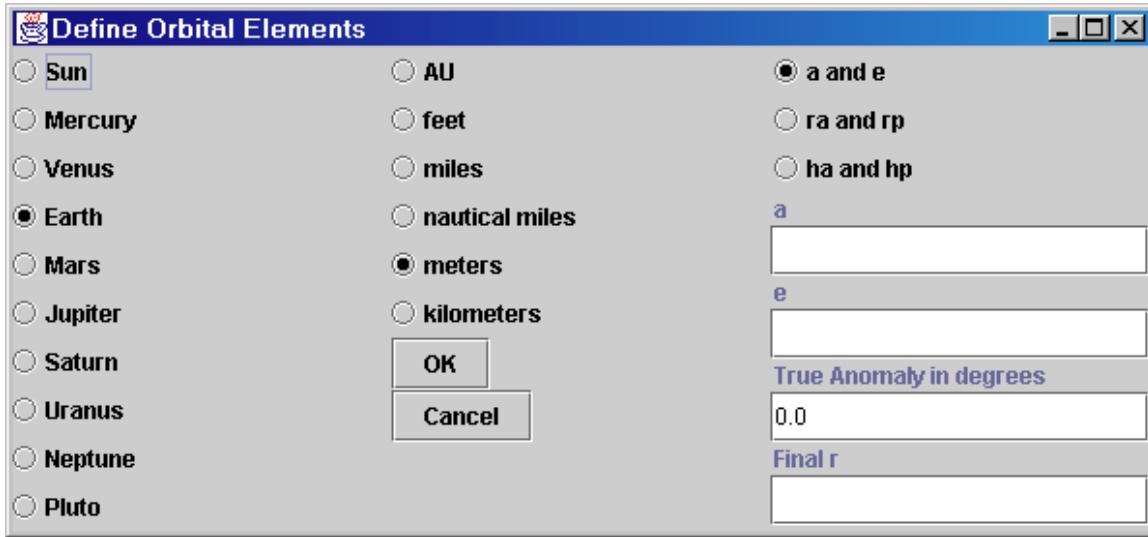


Figure 3: Orbit Definition Window

D. Thrust Definition Window

The thrust definition window is used to define the thrust profile over the course of the trajectory. The thrust profile is based on the true anomaly or mean anomaly of the spacecraft in orbit. Multiple thrust definitions can be pieced together so long as all 360 degrees are defined. The thrust direction and magnitude are linearly interpolated between a starting and ending value specified by the user with the range of true (mean) anomaly. The starting and ending thrust direction can be relative to the velocity or the position vector. The angle is measured counter-clockwise from the vector, so -5 degrees relative to the velocity vector points slightly away from the Earth in a circular orbit. Likewise, +90 degrees from the position vector (defined from the center of the orbital body to the spacecraft) is along the velocity vector for a circular orbit. The starting and ending magnitude of thrust can be specified in N, mN, or lbs. The thrust definition window is shown in Figure 4.

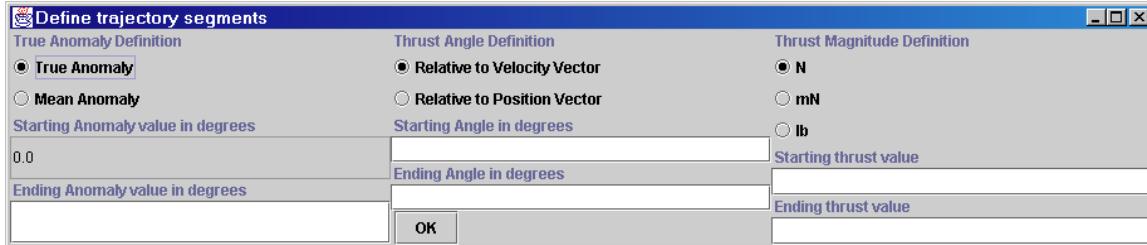


Figure 4: Thrust Definition Window

E. Vehicle Definition Window

The vehicle definition window defines the vehicle mass and the propulsion system's specific impulse (I_{SP}) in seconds. The vehicle mass can be defined in kg, lb_m, or slugs. The default values are 1000 kg and 3000 sec. The vehicle definition window is shown in Figure 5.

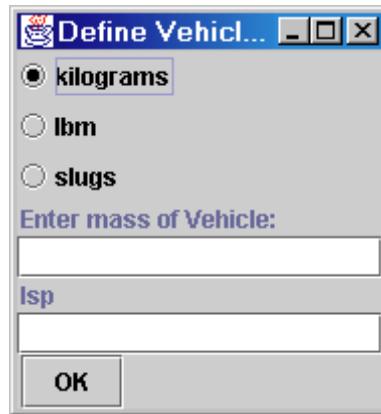


Figure 5: Vehicle Definition Window

III. Calculations

The program utilizes an iteration loop to calculate position, velocity, acceleration, and mass of the vehicle, as well as the orbital parameters and the time elapsed. This data is used to update the animation of the orbit and the orbital data listed in the main window.

A. Acceleration calculation

The acceleration due to gravity is calculated based on the distance from the center of the body using Equation 1 [1].

$$\vec{a} = \square \frac{\square}{r^2} \hat{r} \quad (1)$$

Thrust is determined from the user-defined thrust profile. True (mean) anomaly of the orbit is calculated and used to determine the applicable region of the thrust profile. A linear interpolation between the minimum and maximum thrust is used to determine the appropriate thrust magnitude. Similarly, the angle relative to the position vector or the velocity vector as specified by the user is determined from a linear interpolation between the minimum and maximum angle specified. The magnitude and angle are then used to determine the vector for thrust. Equation 2 shows this interpolation scheme for the angle using the true anomaly, \square , as the independent variable, and Equation 3 shows the interpolation for the thrust magnitude, T , using the mean anomaly, M , as the independent variable. Which variable is actually used as the independent variable is determined by the user's selection in the thrust definition window.

$$Angle = \frac{Angle_{final} - Angle_{initial}}{\square_{final} - \square_{initial}} (\square - \square_{initial}) + Angle_{initial} \quad (2)$$

$$T = \frac{T_{final} - T_{initial}}{M_{final} - M_{initial}} (M - M_{initial}) + T_{initial} \quad (3)$$

Newton's second law of motion is shown in Equation 4. Using the chain rule leads to Equation 5. Typically this is reduced to Equation 6 with the assumption that mass is constant. This is normally a very bad assumption to make for solving problems involving rockets where the expulsion of propellant (and therefore a change in mass) is used to produce the thrust. In this case, however, the time step used is so small that the instantaneous acceleration follows the result of Equation 6. This method is presented in

Reference 2. The mass of the vehicle will be calculated for the next time step, so the effect of the changing mass is taken into account by the numerical integration scheme.

$$\vec{T} = \frac{d(m\vec{v})}{dt} \quad (4)$$

$$\vec{T} = \frac{d(m)}{dt} \vec{v} + \frac{d(\vec{v})}{dt} m \quad (5)$$

$$\vec{T} = m\vec{a} \quad (6)$$

The total acceleration on the vehicle is the vector sum of the acceleration due to gravity and the acceleration from the thrust imparted by the vehicle's engines.

B. Velocity and Position Calculations

The velocity is calculated by integrating the acceleration as shown in Equation 7. The numerical approximation uses a finite time step, Δt , and approximates the relationship as shown in Equation 8.

$$\vec{v} = \int \vec{a} dt \quad (7)$$

$$\vec{v} = \vec{v}_0 + \vec{a}\Delta t \quad (8)$$

Similarly, the position is calculated by integrating the velocity as shown in Equation 9, and the numerical approximation is shown in Equation 10.

$$\vec{r} = \int \vec{v} dt \quad (9)$$

$$\vec{r} = \vec{r}_0 + \vec{v}\Delta t \quad (10)$$

C. Mass and Time Calculations

The mass flow rate of propellant expelled is determined from the specific impulse and the thrust as defined in Equation 11 where g_c is the gravitational acceleration at sea level on the Earth. The mass of the vehicle decreases as a function of time according to Equation 12. The numerical approximation of the final mass is given in Equation 13. Note that in these equations, \dot{m} is the *propellant* mass flow rate and not the rate of change of mass of the vehicle. The rate of change of mass of the vehicle is the negative of this number.

$$I_{SP} = \frac{T}{\dot{m}g_c} \quad (11)$$

$$m = m_0 - \int \dot{m} dt \quad (12)$$

$$m = m_0 - \bar{m}t \quad (13)$$

The final time is simply the initial time plus the time step as shown in Equation 14.

$$t = t_0 + \Delta t \quad (14)$$

D. Determination of the Time Step

During any particular iteration, the time step is initially the time step from the previous iteration. The above numerical approximations are used to calculate the final position and velocity of the vehicle. They are then calculated again using half that time step two times. If there were no errors due to the approximations, the two solutions should come up with the same answer, since the total time elapsed for both cases is the same. Comparing the two solutions gives an estimate of the error in the calculations. If

this difference exceeds the tolerance (1×10^{-8} times the smaller of the initial periapsis or final distance from the center of the body), the time step is reduced. If the difference is less than half of the tolerance, the time step is increased. In this way, the magnitude of the error is always kept relatively constant.

E. Determination of the Orbital Elements

The orbital elements are determined from the position and velocity vectors of the spacecraft and the gravitational parameter, μ , of the body which is being orbited. The orbital elements are not used to determine spacecraft position or motion, but they are used to determine the true and mean anomaly for the thrust profile. The orbital mechanics equations all come from Reference 1.

The energy of the orbit, E , is calculated from the magnitude of the velocity vector and the magnitude of the position vector as shown in Equation 15.

$$E = \frac{1}{2} v^2 \mu \frac{\mu}{r} \quad (15)$$

The specific angular momentum vector, \mathbf{h} , is calculated by taking the cross product of the position and velocity vectors as shown in Equation 16. Only the magnitude of \mathbf{h} is used in the calculations.

$$\vec{h} = \vec{r} \times \vec{v} \quad (16)$$

The flight path angle, γ , is determined from Equation 17. Since the cosine is symmetric about the y-axis, the arc-cosine only determines the value from 0 to 180 degrees. The sign of γ is the same as the sign as $\vec{r} \cdot \vec{v}$.

$$h = r v \cos \gamma \quad (17)$$

The semi-major axis, a , is determined from Equation 18.

$$a = \frac{\mu}{2E} \quad (18)$$

The semi-latus rectum, p , of the orbit is found by Equation 19.

$$p = \frac{h^2}{\mu} \quad (19)$$

The eccentricity, e , can be found then from Equation 20. Numerical inaccuracies can cause p/a to be greater than 1.0, even though this is a non-physical result. If this is the case, e is set to 0.0.

$$e = \sqrt{1 - \frac{p}{a}} \quad (20)$$

The periapsis, r_p , and apoapsis, r_a , are found from Equations 21 and 22, respectively.

$$r_p = a(1 - e) \quad (21)$$

$$r_a = a(1 + e) \quad (22)$$

From the definition of specific angular momentum, the velocity at periapsis, v_p , and the velocity at apoapsis, v_a , can be found from equations 23 and 24.

$$v_p = \frac{h}{r_p} \quad (23)$$

$$v_a = \frac{h}{r_a} \quad (24)$$

The true anomaly, \square , is determined by rearranging equation 25 to solve for \square as shown in Equation 26. If $\vec{r} \bullet \vec{v}$ is negative, then the sign of \square is negative. Due to numerical inaccuracies, if the argument of the arc cosine in Equation 26 is greater than 1.0, \square is set to 0 degrees, and if it is less than -1.0, \square is set to 180 degrees. Note that \square is not defined for a circular orbit, so for any orbit with a sufficiently small e , \square is set to 0.

$$r = \frac{p}{1 + e \cos \square} \quad (25)$$

$$\square = \cos^{-1} \frac{r}{e} \quad (26)$$

The period is calculated by Equation 27.

$$Period = \frac{2\square}{\sqrt{\square}} a^{\frac{3}{2}} \quad (27)$$

The calculation of mean anomaly, M , depends on the type of orbit. The mean anomaly, similar to the true anomaly, is not defined for a circular orbit, and is therefore set to 0 in this case. The equations for calculating mean anomaly are taken from Reference 3. For an elliptical orbit where $0 < e < 1$, M is given by Equation 28 where the parameter u is given by Equation 29. For a hyperbolic orbit, where $e > 1$, M is given by Equation 30 where the parameter u is given by Equation 31.

$$M = u \square e \sin u \quad (28)$$

$$u = \cos^{-1} \frac{e + \cos \square}{1 + e \cos \square} \quad (29)$$

$$M = e \sinh u \Box u \quad (30)$$

$$u = \cosh^{-1} \frac{e + \cos \theta}{1 + e \cos \theta} \quad (31)$$

IV. Java Programming

A. Object Oriented Programming

The Java programming language uses object-oriented programming. The code is written using multiple *classes*, which define how the data is used and manipulated. An *object* is a collection of data belonging to a given class. For example, one class used in the Trajectory Animator is the MyVector class. (Java has its own Vector class which is not related to 2D vectors) In this class, there are class *methods* that can manipulate the data contained in the MyVector object. Position and velocity are two objects of the MyVector class used in the Trajectory Animator. The MyVector class contains methods that can be used to calculate the dot product, cross product, or addition of two MyVector objects as well as several other useful functions. This style of programming allows the programmer to create a library of useful tools (the classes) that are easy to manipulate to produce the desired outcome. The source code for all of the Java classes is presented in Appendix A.

B. Animation

Java makes graphical user interfaces extremely user friendly, including animation. In order to animate the orbits, it was necessary to convert the orbital coordinates to pixels on the screen. The geometric modeling techniques, a series of affine transformations, are given in Reference 4. First, the center of the orbital picture is moved to the origin. Then the coordinates are scaled to the proper dimension (which can later be changed by a

scaling factor set by the user with the zoom slider bar). Next the picture is translated so that the center of the picture is in the center of the window, which is how it is displayed. The same transformations are applied to the orbital body, the initial orbit, and the final position.

V. Verification

A. Orbital Progression

Simply animating an orbit without any thrust applied demonstrates the accuracy of the orbital progression. The green orbit in the trajectory animation window is determined by plotting 1000 points of the orbit based on the initial orbital parameters. For orbits that are not highly elliptical, the error is typically so small that it is hard to distinguish any difference between the two paths. The change in the orbital parameters in the main window is a better indication of the error. If there were no error, the orbital parameters would remain constant. Typically, the specific angular momentum does not change at all. For the case of $a = 15,000$ km about the Earth and $e = 0.3$, the Energy is 1% off after 7 complete orbits.

B. Thrust Application

Thrust can be applied in any magnitude desired. Typically thrust is considered to be “high thrust” which for calculations can be assumed to be nearly instantaneous or “low thrust” where the thrust is applied continuously over a long period of time, usually with a much more efficient (higher I_{SP}) engine. The trajectory animator can simulate both types of thrust profiles, although defining the true anomaly limits for a high thrust case is not intuitive.

1. High Thrust

As a test case, a sample orbit was created with $a = 8000$ km and $e = 0.1$. The main window after this setup is shown in Figure 6. The orbital elements shown agree with those calculated using Equations 13 through 25.

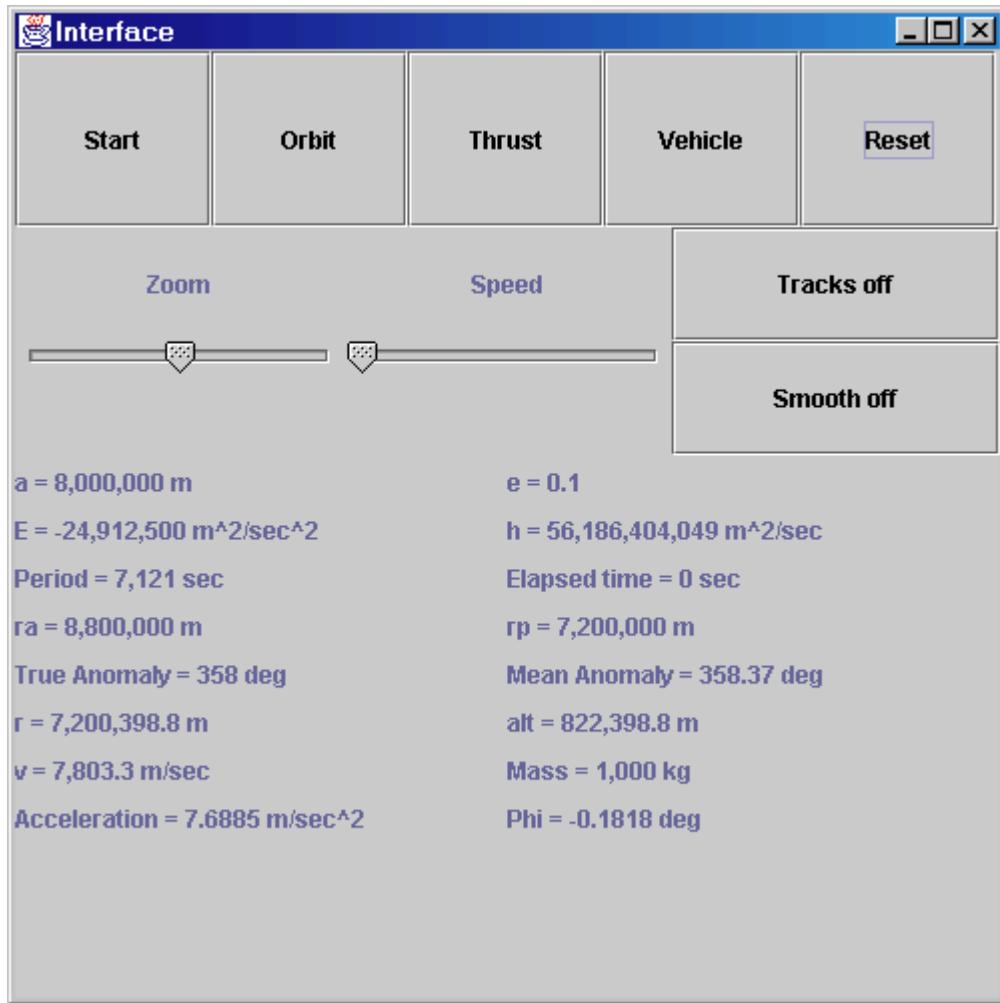


Figure 6: High Thrust Test Case Initial Status

A thrust of 10,000 N from an engine with an I_{sp} of 460 sec is used in this test case. The engine is on from a true anomaly of 358 to 002 degrees. (Note that in running the program, since the thrust definition is always started at 0 degrees true or mean anomaly, this is set in 3 steps: 0 to 2 degrees with 10,000 N of thrust, 2 to 358 degrees

with zero thrust, and 358 to 360 degrees with 10,000 N of thrust. The orbit is started at a true anomaly of 358 degrees. The thrust angle is defined as 0 degrees relative to the velocity vector.) The results of this simulated burn are shown in Figure 7. The total elapsed time is only 60 seconds, so this is a good approximation of an instantaneous burn for ΔV calculations. The calculated change in velocity over this time is 8448.7 m/sec – 7803.3 m/sec or 645.4 m/sec. The calculated change in mass is from 1000 kg to 866.3 kg. The ratio of initial mass to final mass is 1.154. Using Equation 32, the rocket equation, the calculated ratio is 1.154, which is equivalent to the calculated value.

$$\frac{m_i}{m_f} = e^{\frac{\Delta V}{g_c I_{SP}}} \quad (32)$$

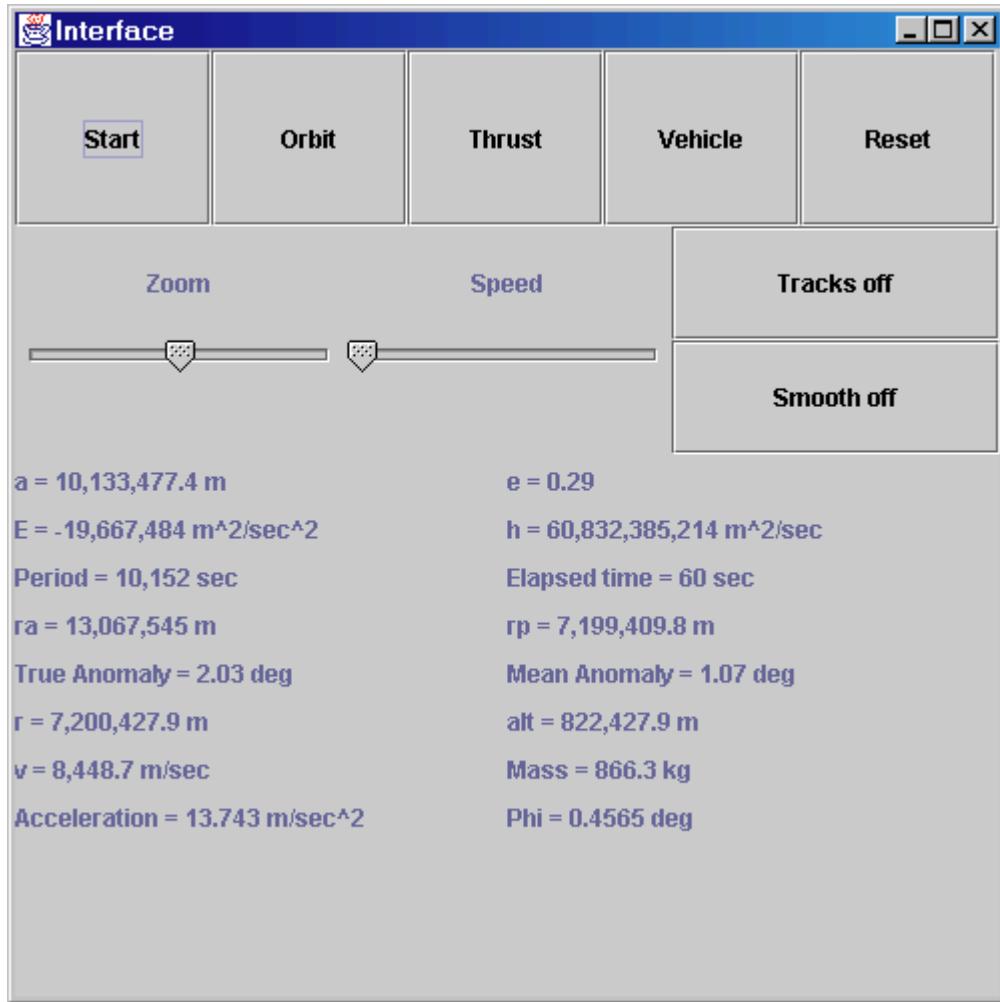


Figure 7: High Thrust Test Case Final Status

If the above test case is repeated using a very high I_{SP} (1×10^9 sec), after a 60 second burn time, the velocity is 8404 sec (see Figure 8). This is equivalent to a 600.7 m/sec ΔV . The predicted ΔV using Equation 33 is 600 m/sec. Equation 33 is simply the application of Equation 8, substituting thrust/mass for acceleration. In this case, since the mass is constant, it is permissible to use this equation without the instantaneous acceleration assumption using very small time steps.

$$\Delta V = \frac{\Delta t * T}{m} \quad (33)$$

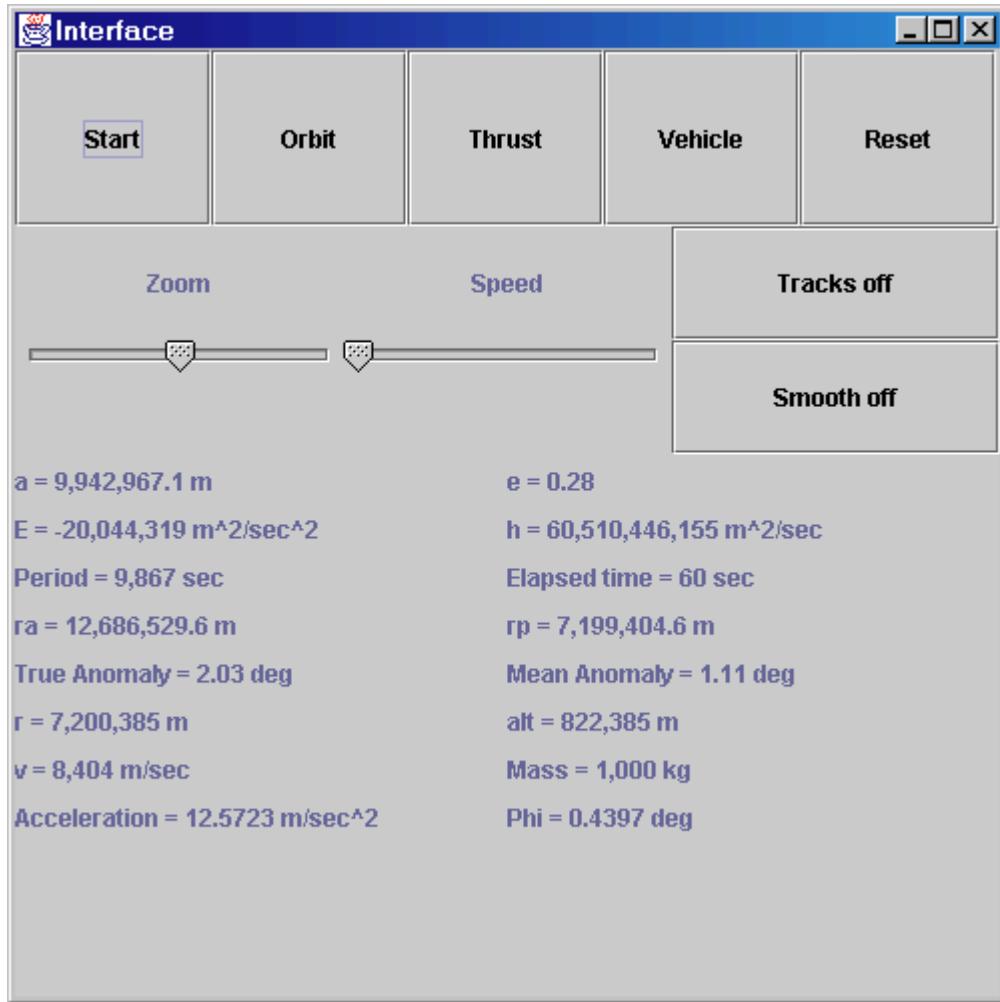


Figure 8: High Thrust, Constant Spacecraft Mass Test Case

2. Low Thrust

The output of the Trajectory Animator was compared to two 4th order Runge-Kutta simulations of a low thrust trajectory [5,6]. These simulations included the oblateness of the Earth, which the Trajectory Animator does not. In the simulations, a 10 kg spacecraft uses an engine with 1 N of thrust at a specific impulse of 3000 seconds. The simulation required the spacecraft to move from a 7015 km circular orbit to a final distance of 42235 km. In the simulation, the thrust was vectored 5 degrees clockwise from the velocity vector for the first 8068 seconds, and then was vectored 2 degrees

clockwise from the velocity vector for the remainder of the flight. In the simulation, the elapsed time was 45,875 seconds and the final mass of the spacecraft was 8.44 kg. Figure 9 shows a plot of this solution (1 DU = 6378 km).

This condition was simulated using the Trajectory Animator. The initial conditions are shown in Figure 10. The thrust was defined to be 1 N, -5 degrees relative to the velocity vector for all 360 degrees of the true anomaly. Once 8068 seconds had passed, the simulation was paused, and the thrust redefined with the only change being thrust applied -2 degrees from the velocity vector (Trajectory Animator references angles counter-clockwise).

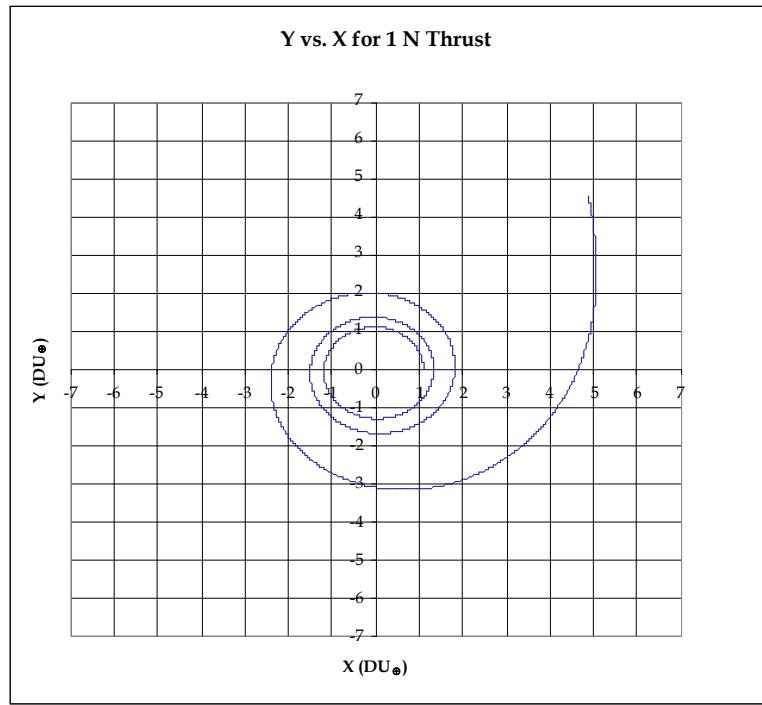


Figure 9: 4th Order Runge-Kutta Solution

The final status of the orbital propagation is shown in Figure 11, and the animation of the orbit is shown in Figure 12. The animation is almost identical to that in Figure 9. A comparison of the results is shown in Table 1. As can be seen, the results are very close. The difference in calculated time was only 591 seconds, which is only 1.3% of the Runge-Kutta solution value of 45,409 sec.

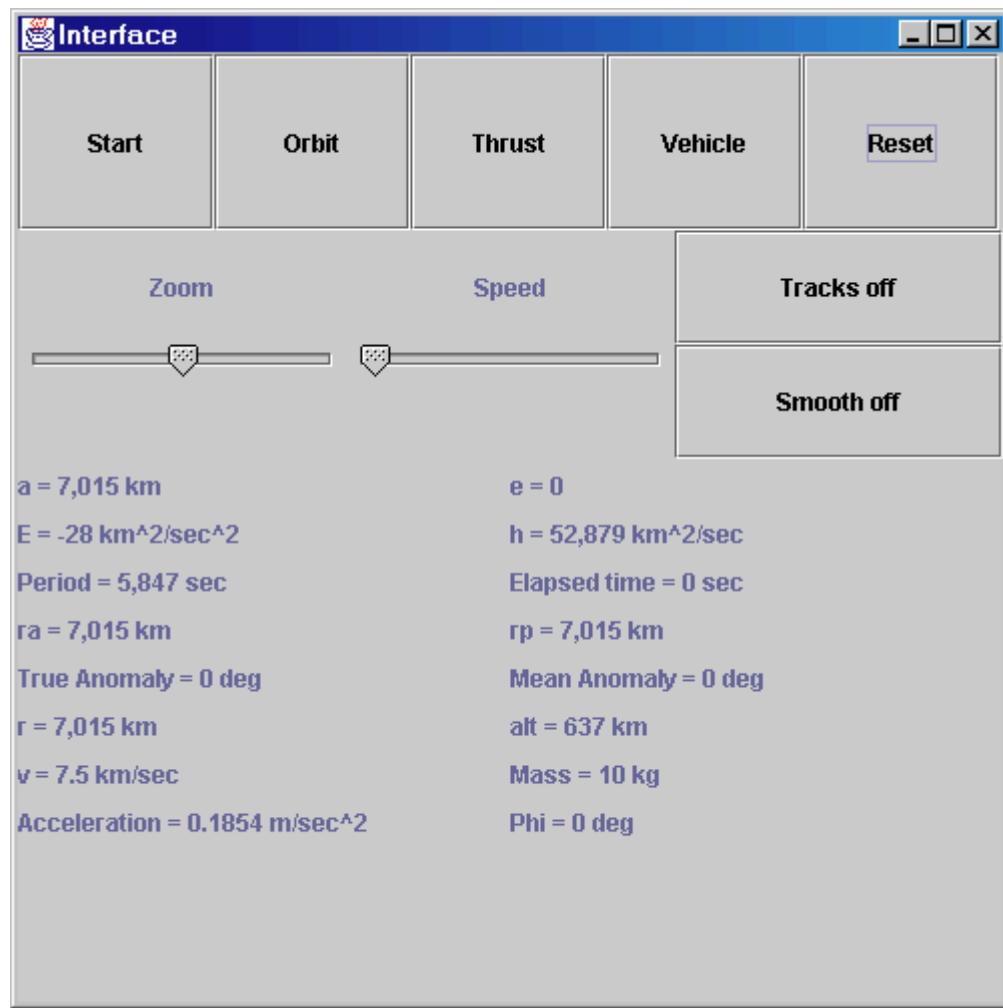


Figure 10: 1 N Low Thrust Test Case Initial Settings

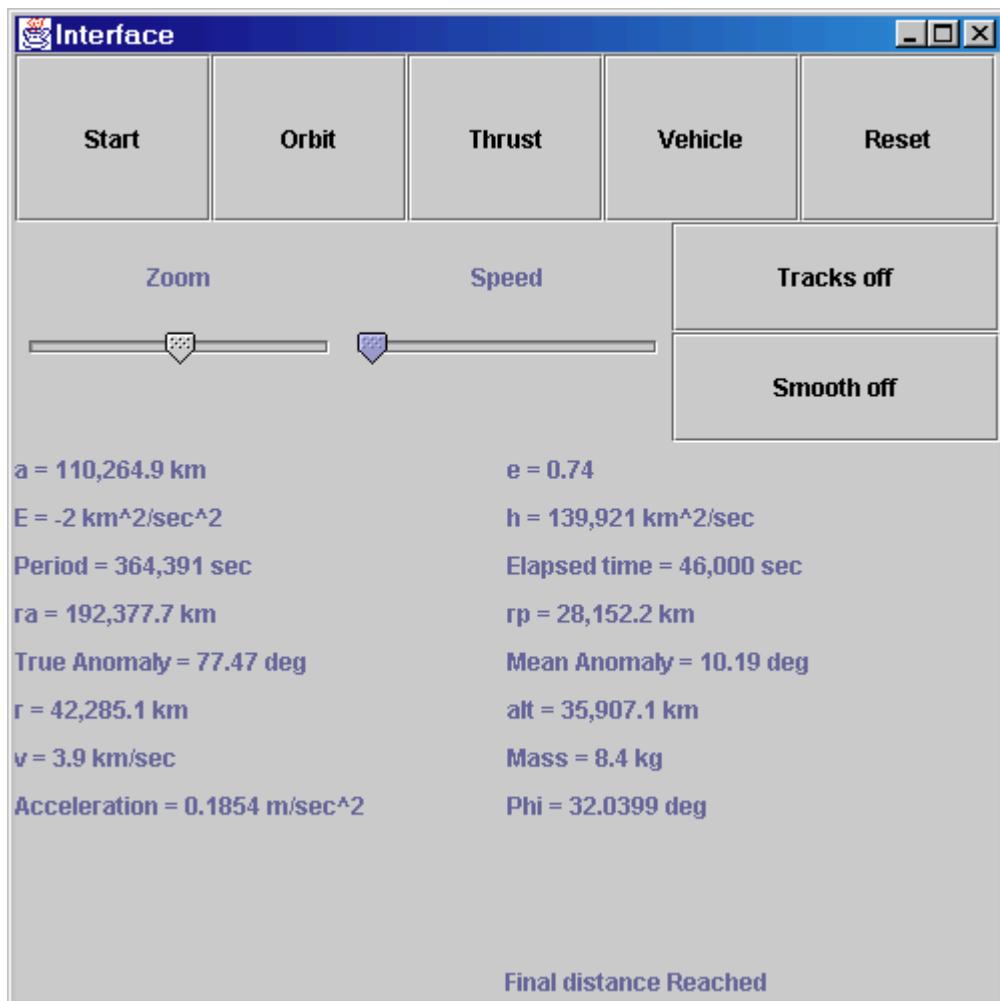


Figure 11: 1 N Low Thrust Test Case Final Settings

Table 1: Comparison Between Runge-Kutta Code and Trajectory Animator, 1 N Thrust

| | 4 th Order Runge-Kutta | Trajectory Animator |
|--------------------|-----------------------------------|---------------------|
| Final Mass | 8.43 kg | 8.4 kg |
| Final Eccentricity | 0.76 | 0.74 |
| Propagation Time | 45,409 sec | 46,000 sec |

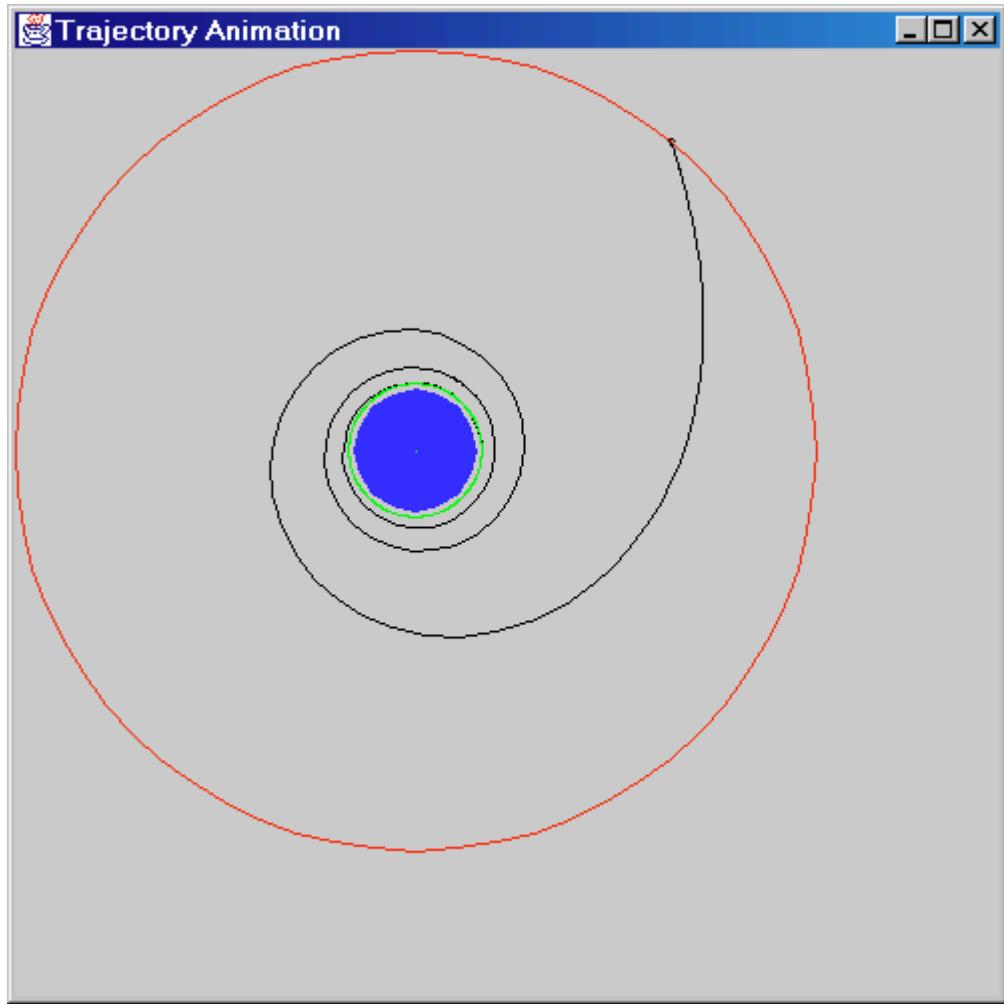


Figure 12: 1 N Low Thrust Test Case Animation Window

Another case using the same conditions except 0.01 N of thrust was also run for comparison. The comparison of these results to the higher order solution is presented in Table 2. In this case, the difference in the time of flight is 3718 seconds, which is only 0.1 % of the Runge-Kutta solution of 4,150,840 sec. The final orbital parameters are shown in Figure 13, and the animation is shown in Figure 14. In order to run this case more quickly, the data was not smoothed, and this can be seen in Figure 14, but it does not affect the accuracy of the calculations.

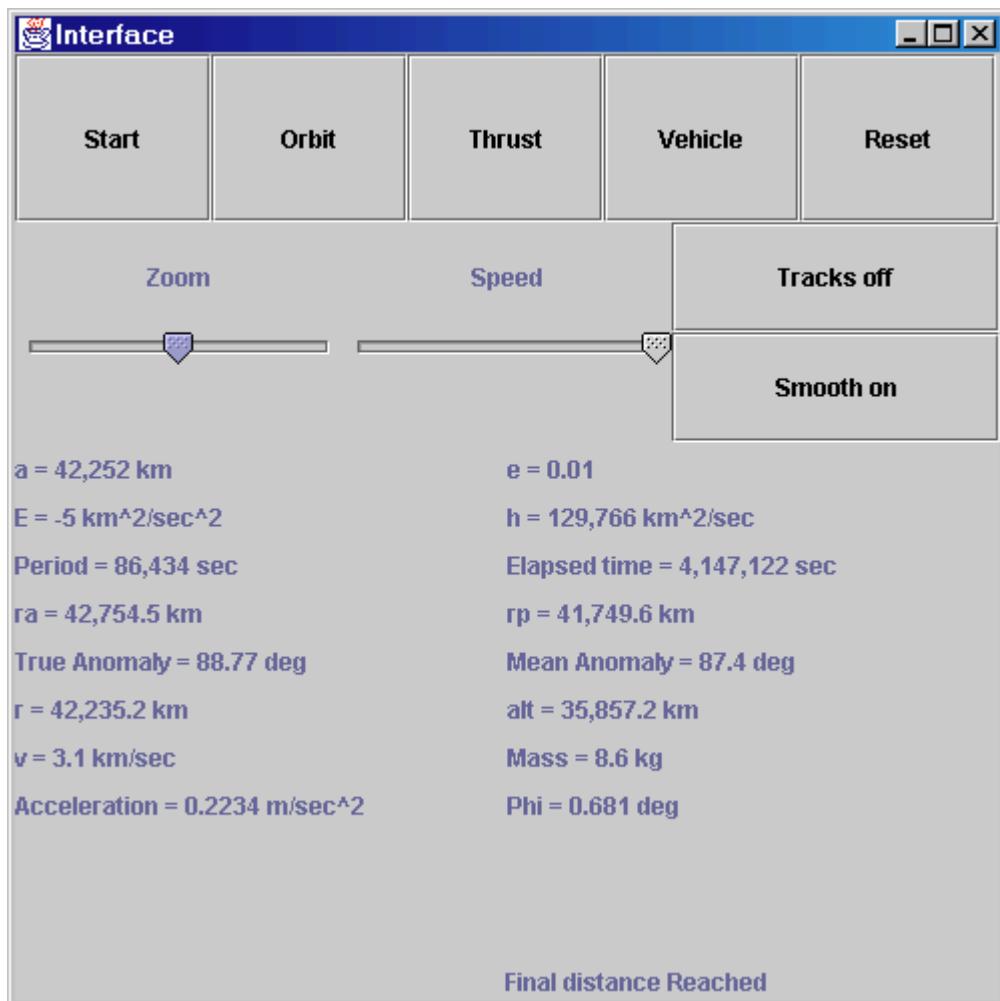


Figure 13: 0.01 N Low Thrust Test Case, Final Results

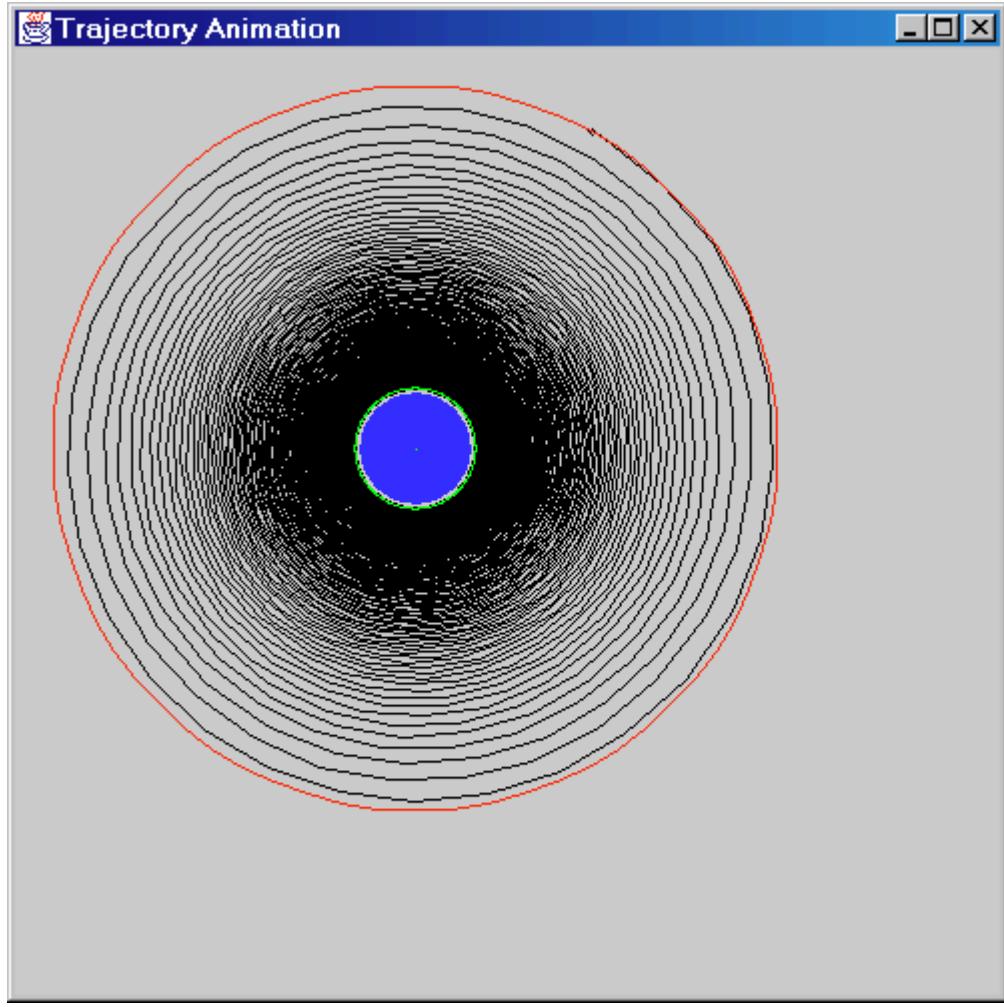


Figure 14: 0.01 N Low Thrust Test Case Animation Window

Table 2: Comparison Between Runge-Kutta Code and Trajectory Animator, 0.01 N Thrust

| | 4 th Order Runge-Kutta | Trajectory Animator |
|--------------------|-----------------------------------|---------------------|
| Final Mass | 8.58 kg | 8.6 kg |
| Final Eccentricity | 0.01 | 0.01 |
| Propagation Time | 4,150,840 sec | 4,147,122 |

VI. Limitations

The Trajectory Animator works best for orbits that are not highly elliptical. All orbits will see errors near the apoapsis where the acceleration due to gravity is smaller. This error becomes pronounced for orbits with an eccentricity greater than about 0.6.

Small numerical errors coupled with the small acceleration can accumulate and make a significant change to the orbit. This is seen as the apoapsis in subsequent orbits is reduced without any thrust applied. If an orbit is animated without any thrust applied, the user can see after several orbits the magnitude of the error in the orbit.

For thrust definition, the true or mean anomaly must be specified between 0 and 360 degrees. For an elliptical orbit, this does not pose a problem, as all points on the ellipse are contained in this range. For a hyperbolic orbit, however, the mean anomaly extends from negative infinity to positive infinity, and the true anomaly is bounded by the asymptotes, so it does not encompass all of the 360 degrees. If a hyperbolic orbit is used, the thrust should be defined with true anomaly, and in the thrust definition, values must be supplied for the values of true anomaly that will never be reached, even though the program will not make use of them.

VII. References

1. Bate, Roger R., Mueller, Donald D., White, Jerry E., *Fundamentals of Astrodynamics*, New York, NY, Dover Publications, Inc., 1971.
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3. Hale, Francis J., *Introduction to Space Flight*, Prentice-Hall, 1994.
4. Mortenson, M. E., *Geometric Modeling*, 2nd Ed., John Wiley & Sons, 1997.
5. Nelson, Doug, “A Program to Propagate a Low Thrust Trajectory Using Cowell’s Method”, 2000.
6. St. Germain, Brad, FORTRAN Low-thrust trajectory code

Appendix A: Source Code

```

import java.awt.event.ActionListener;
public class Interface extends JApplet implements ActionListener
{
    /*
     * Interface.java
     * Created on September 14, 2001, 4:07 PM
     */
    /**
     * @author John E. Weglian
     * @version 1.0
     */
    import javax.swing.JFrame;
    import javax.swing.JButton;
    import javax.swing.JPanel;
    import javax.swing.JLabel;
    import javax.swing.JComboBox;
    import javax.swing.JRadioButton;
    import javax.swing.ButtonGroup;
    import javax.swing.JTextField;
    import javax.swing.BoxLayout;
    import javax.swing.Applet;
    import javax.swing.Timer;
    import javax.swing.JSlider;
    import javax.swing.event.ChangeListener;
    import javax.swing.event.ChangeEvent;
    import javax.swing.SwingConstants;
    import java.awt.event.ActionEvent;
    import java.awt.Container;
    import java.awt.GridLayout;
    import java.awt.GeneralPath;
    import java.awt.geom.AffineTransform;
    import java.awt.Graphics;
    import java.awt.Graphics2D;
    import java.awt.Color;
    import java.awt.Dimension;
    import java.awt.GridBagLayout;
    import java.awt.GridBagConstraints;
    import java.awt.Shape;
    import java.util.ArrayList;

```

```

private JLabel finalRLLabel;
private JButton smoothButton;
private JLabel positionLabel;
private JButton aeButton;
private JLabel massLabel;
private JButton rarpButton;
private JLabel IspLabel;
private JButton hahpButton;
private JLabel defineVehicleErrorMessage;
private JButton sunButton;
private JButton mercuryButton;
private JLabel massValueLabel;
private JButton venusButton;
private JButton statusLabel;
private JButton marsButton;
private JButton earthButton;
private JButton zoomLabel;
private JButton jupiterButton;
private JButton timeLabel;
private JButton saturnButton;
private JButton velocityLabel;
private JButton uranusButton;
private JButton altitudeLabel;
private JButton neptuneButton;
private JButton accelerationLabel;
private JButton plutoButton;
private JButton pauseButton;
private JButton ftButton;
private JButton defineOrbitButton;
private JButton miButton;
private JButton defineThrustButton;
private JButton rmButton;
private JButton restartButton;
private JButton mButton;
private JButton defineOrbitOKButton;
private JButton AUButton;
private JButton defineOrbitCancelButton;
private JButton kmButton;
private JButton defineVehicleButton;
private JButton kgButton;
private JButton defineVehicleOKButton;
private JButton lbmButton;
private JButton trackButton;
private JButton slugButton;

```

```

private JTextField _aorTextField;
private JTextField _eorTextField;
private JTextField _trueAnomalyTextField;
private JTextField _finalRTextField;
private JTextField _massTextField;
private JTextField _IspTextField;
private JSlider _speedSlider;
private JSlider _zoomSlider;
private GeneralPath _trajectoryPath;
private GeneralPath _plotPath;
private GeneralPath _bodyPath;
private GeneralPath _initialOrbitPath;
private GeneralPath _finalOrbitPath;
private GeneralPath _satellite;
private GeneralPath _initBodyPath;
private GeneralPath _initInitialOrbitPath;
private GeneralPath _initFinalOrbitPath;
private AffineTransform _transform;
private Timer _timer;
private Orbit _currentOrbit;

private Orbit _initialOrbit;
private Color _bodyColor;
private NumberFormat _format;
private NumberFormat getInstance();
private int _dimension;
private int _textBoxLength;
private int _body;
private int _selection;
private int _delay;
private int _errorStatus;
private double _scaleFactor;
private double _aorValue;
private double _eорValue;
private double _factor;
private double _trueAnomalyValue;
private double _finalRVvalue;
private double _centerX;
private double _centerY;
private double _max;
private double _massFactor;
private double _massValue;
private double _IspValue;

private Orbit = null;
private Color = Color.blue;
private NumberFormat = NumberFormat.getInstance();
private int = 400;
private int = 10;
private int = 3;
private int = 0;
private int = 500;
private int = 0;
private double = 1.0;
private double = 0.0;
private double = 1.0;
private double = 0.0;
private double = 0.0;

```

```

private double           _initialMass;
private double           _initialMax;
private double           _initialMin;
private boolean          _paused;
private boolean          _aorSet;
private boolean          _eorSet;
private boolean          _trueAnomalySet;
private boolean          _finalRSet;
private boolean          _massSet;
private boolean          _IspSet;
private boolean          _plotTracks;
private boolean          _smooth;
private String           _lengthUnitString;
private String           _massUnitString;
private String           _kgUnitString;
private Trajectory        _trajectory;
private PlotPanel         _trajectoryPanel;
private static final double[] _radius = {696000.0e3,
                                         2487.0e3, 6187.0e3,
                                         6378.0e3, 3380.0e3, 71370.0e3, 60400.0e3, 23530.0e3,
                                         22320.0e3, 7016.0e3};

/** Creates new Interface */
public Interface() {
}

public Interface(Container contentPane)
{
    contentPane.setLayout(new BoxLayout(contentPane,
                                       BoxLayout.Y_AXIS));
    contentPane.add(_buttonPanel);
    contentPane.add(_sliderPanel);
    contentPane.add(_vectorPanel);
    contentPane.add(_orbitParametersPanel);
    contentPane.add(_buttonPanel);
    contentPane.add(_sliderPanel);

    contentPane.setLayout(new BoxLayout(contentPane,
                                       BoxLayout.Y_AXIS));
    contentPane.add(_buttonPanel);
    contentPane.add(_sliderPanel);
    contentPane.add(_vectorPanel);
    contentPane.add(_orbitParametersPanel);
    contentPane.add(_buttonPanel);
    contentPane.add(_vectorPanel);
    contentPane.add(_statusLabel);
    contentPane.add(_trajectoryFrame);
    contentPane.add(_trajectoryPanel);
    contentPane.setDoubleBuffered(true);

    trajectoryFrame.getContentPane().add(_trajectoryPanel);
    trajectoryFrame.setSize(500, 500);
    trajectoryFrame.show();
    setupTimer();
}

public void init()
{
    Container contentPane = getContentPane();
    setDefaults();
    _trajectoryPanel = createTrajectoryPanel();
    _orbitParametersPanel = createOrbitParametersPanel();
    _vectorPanel = createVectorPanel();
    _buttonPanel = createButtonPanel();
    _sliderPanel = createSliderPanel();

    contentPane.setLayout(new BoxLayout(contentPane,
                                       BoxLayout.Y_AXIS));
    contentPane.add(_buttonPanel);
    contentPane.add(_sliderPanel);
    contentPane.add(_vectorPanel);
    contentPane.add(_orbitParametersPanel);
    contentPane.add(_buttonPanel);
    contentPane.add(_vectorPanel);
    contentPane.add(_statusLabel);
    contentPane.add(_trajectoryFrame);
    contentPane.add(_trajectoryPanel);
    contentPane.setDoubleBuffered(true);

    trajectoryFrame.getContentPane().add(_trajectoryPanel);
    trajectoryFrame.setSize(500, 500);
}

```

```

    _trajectoryFrame.show();
    _setupTimer();
}

private void setupTimer()
{
    _timer = new Timer (_delay, this);
    _timer.setInitialDelay(0);
    _timer.setCoalesce(true);
}

private void setDefaults ()
{
    Interpolation defaultSegment = new Interpolation (true,
true, 0.0, 360.0, 0.0, 0.0, 0.0);
    ArrayList segment = new ArrayList();
    segment.add (defaultSegment);
    trajectory = new Trajectory (3, 6778.0e3, 7000.0e3,
segment);
    currentOrbit = trajectory.getOrbit();
    initialOrbit = _currentOrbit;
    bodyPath = trajectory.getBdyPath();
    initialOrbitPath = trajectory.getInitialOrbitPath();
    finalOrbitPath = trajectory.getFinalOrbitPath();
    trajectory.initializePath();
    trajectoryPath = trajectory.getPath();
    satellite = trajectory.getSatellite();
    scale();
    initialMass = 1000.0;
    trajectory.setMass (_initialMass);
    trajectory.setSp (3000.0);
}

private JPanel createSliderPanel ()
{
    JPanel panel = new JPanel();
    panel.setLayout(new GridLayout(2, 3));
    _zoomLabel = new JLabel("Zoom");
    _panel.setHorizontalTextPosition(SwingConstants.CENTER);
    panel.add(_zoomLabel);
    _speedLabel = new JLabel("Speed");
    _panel.setHorizontalTextPosition(SwingConstants.CENTER);
    panel.add(_speedLabel);
    _trackButton = new JButton ("Tracks off");
    _trackButton.addActionListener (new ActionListener () {
        public void actionPerformed (ActionEvent e)
        {
            if (_plotTracks)
            {
                plotTracks = false;
                _trajectoryPanel.repaint ();
                _trackButton.setText ("Tracks on");
            }
            else
            {
                plotTracks = true;
                _trajectoryPanel.repaint ("Tracks off");
                _trackButton.setText ("Tracks on");
            }
        }
    });
    panel.add (_trackButton);

    zoomSlider = new JSlider (0, 100, 50);
    zoomSlider.addChangeListener (new ChangeListener () {
        public void stateChanged (ChangeEvent e)
        {
            double value =
Math.exp(0.1*(double)_zoomSlider.getValue());
            scaleFactor = Math.exp(5.0)/value;
            newScale();
        }
    });
    panel.add (_zoomSlider);

    speedSlider = new JSlider (0, 1000, 500);
    speedSlider.addChangeListener (new ChangeListener () {
        public void stateChanged (ChangeEvent e)
        {
            int value = _speedSlider.getValue();
            delay = 100/_value;
            timer.setDelay(_delay);
            if (_smooth)
                trajectory.setLimit(value/10);
            else
            {
                if (value < 250)
                    _trajectory.setLimit(value/10);
            }
        }
    });
}

```

```

    else
        _trajectory.setLimit(25);
    }
}

panel.add(_speedSlider);
}

smoothButton = new JButton ("Smooth off");
smoothButton.addActionListener (new ActionListener() {
public void actionPerformed (ActionEvent e)
{
    if (_smooth)
    {
        smooth = false;
        int value = _speedSlider.getValue();
        if (value < 250)
            _trajectory.setLimit(value/10);
        else
            _trajectory.setLimit(25);
        _trajectoryPanel.repaint();
        _smoothButton.setText("Smooth off");
    }
    else
    {
        smooth = true;
        int value = _speedSlider.getValue();
        _trajectory.setLimit(value/10);
        _trajectoryPanel.repaint();
        _smoothButton.setText("Smooth on");
    }
}
});

panel.add(_smoothButton);
}

return panel;
}

private JPanel createTrajectoryPanel ()
{
    PlotPanel panel = new PlotPanel ();
    return panel;
}

private JPanel createOrbitParametersPanel ()
{

```

```

+ _lengthUnitString + "2/sec");
+ format.format(_currentOrbit.getTrueAnomaly());
+ " deg");
_meanAnomalyLabel = new JLabel("Mean Anomaly = "
+ format.format(_currentOrbit.getMeanAnomaly())
+ " deg");
format.setMaximumFractionDigits(4);
_accelerationLabel = new JLabel("Acceleration = "
+ format.format(a) + " m/sec^2");
_phiLabel = new JLabel("Phi = "
+ format.toDegrees(_currentOrbit.getPhi()))
+ " deg");

panel.add(_aLabel);
panel.add(_eLabel);
panel.add(_energyLabel);
panel.add(_hLabel);
panel.add(_periodLabel);
panel.add(_timeLabel);
panel.add(_rLabel);
panel.add(_rpLabel);
panel.add(_trueAnomalyLabel);
panel.add(_meanAnomalyLabel);
panel.add(_positionLabel);
panel.add(_altitudeLabel);
panel.add(_velocityLabel);
panel.add(_massLabel);
panel.add(_accelerationLabel);
panel.add(_phiLabel);

return panel;
}

public void updateOrbit()
{
    currentOrbit = _trajectory.getOrbit();
    double r = _currentOrbit.getPosition().getMagnitude();
    double v = _currentOrbit.getVelocity().getMagnitude();
    double a =
        trajectory.getAcceleration().getMagnitude();
    format.setMaximumFractionDigits(0);
    energyLabel.setText("E = "
+ format.format(_currentOrbit.getEnergy() / (_factor * _factor))
+ _lengthUnitString + "2/sec");
    hLabel.setText("h = "
+ format.format(_currentOrbit.getH() / (_factor * _factor))
+ " sec");
    lengthLabel.setText("Period = "
+ format.format(_currentOrbit.getPeriod() + " sec");
    timeLabel.setText("Elapsed time = "
+ format.format(_trajectory.getTime() / _sec));
    format.setMaximumFractionDigits(1);
    aLabel.setText("a = "
+ format.format(_currentOrbit.getA()) / _factor)
+ _lengthUnitString);
    rLabel.setText("ra = "
+ format.format(_currentOrbit.getRa() / _factor)
+ _lengthUnitString);
    rpLabel.setText("rp = "
+ format.format(_currentOrbit.getRp() / _factor)
+ _lengthUnitString);
    positionLabel.setText("r = "
+ format.format(r / _factor)
+ _lengthUnitString);
    massLabel.setText("Mass = "
+ format.format(_trajectory.getMass() / _massFactor)
+ _massUnitString);
    velocityLabel.setText("v = "
+ format.format(v / _factor)
+ _lengthUnitString + "/sec");
    altitudeLabel.setText("alt = "
+ format.format((r - radius[_body]) / _factor)
+ _lengthUnitString);
    eLabel.setText("e = "
+ format.setMaximumFractionDigits(2);
    format.setText("phi = "
+ format.format(_currentOrbit.getPhi()));
    phiLabel.setText("phi = "
+ format.format(_currentOrbit.getPhi())
+ " deg");
    trueAnomalyLabel.setText("True Anomaly = "
+ format.format(_currentOrbit.getTrueAnomaly())
+ " deg");
    meanAnomalyLabel.setText("Mean Anomaly = "
+ format.setMaximumFractionDigits(4);
    accelerationLabel.setText("Acceleration = "
+ format.format(a + " m/sec^2");
    phiLabel.setText("Phi = "
+ format.format(Math.toDegrees(_currentOrbit.getPhi()))
+ " deg");
}

```

```

        }
    }

    private JPanel createVectorPanel ()
    {
        JPanel panel = new JPanel () ;

        return panel;
    }

    private JPanel createButtonPanel ()
    {
        JPanel panel = new JPanel () ;
        panel.setLayout(new GridLayout(1, 4));
        _pauseButton = new JButton ("Start");
        _pauseButton.addActionListener (new ActionListener () 
        {
            public void actionPerformed (ActionEvent e)
            {
                pause();
            }
        });
        panel.add(_pauseButton);

        _defineOrbitButton = new JButton ("Orbit");
        _defineOrbitButton.setToolTipText("Define the initial
        orbital parameters");
        _defineOrbitButton.addActionListener (new
        ActionListener () 
        {
            public void actionPerformed (ActionEvent e)
            {
                if (! _paused)
                    pause();
                defineorbit();
            }
        });
        panel.add(_defineOrbitButton);

        _defineThrustButton = new JButton ("Thrust");
        _defineThrustButton.setToolTipText("Define the thrust
        profile");
        _defineThrustButton.addActionListener (new
        ActionListener () 
        {
            public void actionPerformed (ActionEvent e)
            {
                if (! _paused)
                    pause();
                defineThrust();
            }
        });
        panel.add(_defineThrustButton);

        _defineVehicleButton = new JButton ("Vehicle");
        _defineVehicleButton.setToolTipText("Define the vehicle
        parameters");
        _defineVehicleButton.addActionListener (new
        ActionListener () 
        {
            public void actionPerformed (ActionEvent e)
            {
                if (! _paused)
                    pause();
                defineVehicle();
            }
        });
        panel.add(_defineVehicleButton);

        _restartButton = new JButton ("Reset");
        _restartButton.addActionListener (new ActionListener () 
        {
            public void actionPerformed (ActionEvent e)
            {
                if (! _paused)
                    pause();
                restart();
            }
        });
        panel.add(_restartButton);

        return panel;
    }

    private void defineVehicle()
    {
        _vehicleFrame = new JFrame ("Define Vehicle
        Parameters");
        JPanel panel = new JPanel();
        panel.setLayout (new BoxLayout(panel,
        BoxLayout.Y_AXIS));
        _kgButton = new JRadioButton ("kilograms", true);
        _kgButton.addActionListener (new ActionListener () 
        {

```

```

    {
        public void actionPerformed(ActionEvent event)
        {
            if (_massSet)
            {
                _massValue*=1.0/_factor;
                _massFactor = 1.0;
                _massUnitString = new String (" kg");
            }
        });
        _lbmButton = new JRadioButton ("lbn", false);
        _lbmButton.addActionListener (new ActionListener()
        {
            public void actionPerformed(ActionEvent event)
            {
                if (_massSet)
                    _massValue*=0.4536/_factor;
                _massFactor = 0.4536;
                _massUnitString = new String (" 1bm");
            }
        });
        _slugButton = new JRadioButton ("slugs", false);
        _slugButton.addActionListener (new ActionListener()
        {
            public void actionPerformed(ActionEvent event)
            {
                if (_massSet)
                    _massValue*=14.60592/_factor;
                _massFactor = 14.60592;
                _massUnitString = new String (" slugs");
            }
        });
        ButtonGroup group = new ButtonGroup ();
        group.add(_kgButton);
        group.add(_lbmButton);
        group.add(_slugButton);
        panel.add(_kgButton);
        panel.add(_lbmButton);
        panel.add(_slugButton);
        panel.add(_button);
    });
    massValueLabel = new JLabel ("Enter mass of
Vehicle:");
    panel.add(_massValueLabel);
    massTextField = new JTextField(_textBoxLength);
    _massTextField.setEditable (true);
    _massSet = false;
}
massTextField.addActionListener (new ActionListener()
{
    public void actionPerformed(ActionEvent e)
    {
        try
        {
            String testString =
            _massTextField.getText();
            massValue =
            _massFactor*Double.valueOf(testString).doubleValue();
            if (_massValue < 0.0)
            {
                defineVehicleErrorMessage.setText("Mass must be greater than
0");
            }
            else
            {
                massSet = true;
                _defineVehicleErrorMessage.setText("");
                _ispTextField.requestFocus();
                checkDefineVehicleORStatus ();
            }
        }
        catch (NumberFormatException error)
        {
            _defineVehicleErrorMessage.setText("Enter a
number");
            _massSet = false;
        }
    }
});
_ispLabel = new JLabel ("Isp");
panel.add(_ispLabel);
_ispTextField = new JTextField(_textBoxLength);
_ispTextField.setEditable (true);
_ispSet = false;
_ispTextField.addActionListener (new ActionListener()
{
    public void actionPerformed(ActionEvent e)
    {
        try
    }
});

```

```

    {
        String testString =
        _ispTextField.getText();
        _ispValue =
        Double.valueOf(testString).doubleValue();
        IspSet = true;
        _defineVehicleErrorMessage.setText("");
        _massTextField.requestFocus();
        if (_ispValue < 0.0)
        {
            IspSet = false;
            _ispTextField.setText("");
            _defineVehicleErrorMessage.setText("Isp
must be > 0");
        }
        else
        {
            checkDefineVehicleOkstatus ();
        }
        catch (NumberFormatException error)
        {
            _defineVehicleErrorMessage.setText("Enter a
number");
        }
        panel.add(_ispTextField);
    );
    public void actionPerformed (ActionEvent e)
    {
        massTextField.postActionEvent();
        _ispTextField.postActionEvent();
        if (_massSet && _ispSet)
        {
            initialMass = _massValue;
            trajectory.setMass(_initialMass);
            trajectory.setIsp(_ispValue);
            _vehicleFrame.dispose();
        }
    }
}
private JPanel getOrbitPanel ()
{
    JPanel panel = new JPanel ();
    panel.add(_defineVehicleOKButton);
    vehicleFrame.getContentPane().add(panel);
    _vehicleFrame.pack();
    _vehicleFrame.show();
}

private void updateOrbit ()
{
    IspSet = true;
    _ispTextField.setText("");
    _defineVehicleErrorMessage.setText("");
    _massTextField.requestFocus();
    if (_ispValue < 0.0)
    {
        IspSet = false;
        _ispTextField.setText("");
        _defineVehicleErrorMessage.setText("Isp
must be > 0");
    }
    else
    {
        checkDefineVehicleOkstatus ();
    }
    catch (NumberFormatException error)
    {
        _defineVehicleErrorMessage.setText("Enter a
number");
    }
    panel.add(_ispTextField);
}
private void defineOrbit ()
{
    orbitFrame = new JFrame ("Define Orbital Elements");
    JPanel bodyPanel = getBodyPanel();
    JPanel unitPanel = getUnitPanel();
    JPanel elementPanel = getElementPanel();
    Container contentPane = orbitFrame.getContentPane();
    contentPane.setLayout(new GridLayout(1,3));
    contentPane.add(bodyPanel);
    contentPane.add(unitPanel);
    contentPane.add(elementPanel);
    orbitFrame.pack();
    orbitFrame.show();
}

private JPanel getElementPanel ()
{
    JPanel panel = new JPanel ();
    panel.add(_defineVehicleOKButton);
    vehicleFrame.getContentPane().add(panel);
    _vehicleFrame.pack();
    _vehicleFrame.show();
}

```



```

        _aorTextField.requestFocus();

        catch (NumberFormatException error)
        {
            _errorMessage.setText("Enter a number");
            _aorSet = false;
        }

        if (_aorValue < 0.0)
        {
            _aorSet = false;
            _errorMessage.setText ("a must be positive if e < 1");
            _aorTextField.requestFocus();
        }
        else if (_selection == 1)
        {
            if (_aorValue < _radius [_body])
            {
                _aorSet = false;
                _errorMessage.setText ("ra must be > " + _radius [_body] / _factor);
                _aorTextField.requestFocus();
                if (_eorSet && _aorValue < _eorValue)
                {
                    _aorSet = false;
                    _errorMessage.setText ("ra must be > ep");
                }
                else
                {
                    if (_aorValue < 0.0)
                    {
                        _aorSet = false;
                        _errorMessage.setText ("ha must be > 0");
                        _aorTextField.requestFocus();
                        if (_eorSet && _aorValue < _eorValue)
                        {
                            _aorSet = false;
                            _errorMessage.setText ("ha must be > np");
                            _aorTextField.requestFocus();
                        }
                        checkOKStatus ();
                    }
                    _aorValue < 0.0
                }
            }
            _aorSet = false;
            _errorMessage.setText ("e must be > 1 if a < 0");
            _aorTextField.requestFocus();
        }
        _aorValue < 0.0
    }

    catch (NumberFormatException error)
    {
        _errorMessage.setText("Enter a number");
        _aorSet = false;
    }

    if (_aorValue < 1.0 && _aorSet)
    {
        _aorTextField.requestFocus();
        panel.add(_aorLabel);
        panel.add(_aorTextField);
        _eorTextField = new JTextField (_textBoxLength);
        _eorTextField.setEditable (true);
        _eorTextField.addActionListener (new ActionListener()
        {
            public void actionPerformed(ActionEvent e)
            {
                try
                {
                    String testString =
                    _eorTextField.getText();
                    Double.valueOf (testString).doubleValue ();
                    _errorMessage.setText ("");
                    _eorSet = true;
                    _trueAnomalyTextField.requestFocus ();
                    if (_selection == 0)
                    {
                        if (_eorValue < 0.0)
                        {
                            _eorSet = false;
                            _errorMessage.setText ("e must be > 1.0");
                            _eorTextField.requestFocus ();
                            if (_aorSet)
                            {
                                if (_aorValue > 0.0 && _eorValue >
                                1.0)
                                {
                                    _eorSet = false;
                                    _errorMessage.setText ("e must be > 1 if a < 0");
                                    _eorTextField.requestFocus ();
                                }
                                else if (_eorValue < 1.0 &&
                                _aorValue < 0.0)
                                {
                            }
                        }
                    }
                }
            }
        });
    }
}

```

```

be < 1 if a > 0");
        _errorMessage.setText ("Enter a number");
        _eorSet = false;
    }
}
else if (_selection == 1)
{
    eorValue *= _factor;
    if (_eorValue < _radius [_body])
    {
        eorSet = false;
        _errorMessage.setText ("rp must be
> " + _radius [_body] / _factor);
        _eorTextField.requestFocus ();
        if (_aorSet && _aorValue < _eorValue)
        {
            eorSet = false;
            _errorMessage.setText ("rp must be <
x^a");
            _eorTextField.requestFocus ();
        }
    }
    else
    {
        eorValue *= _factor;
        if (_eorValue < 0.0)
        {
            eorSet = false;
            _errorMessage.setText ("hp must be >
0");
            if (_aorSet && _aorValue < _eorValue)
            {
                eorSet = false;
                _errorMessage.setText ("hp must be <
ha");
                _eorTextField.requestFocus ();
            }
        }
        checkOKStatus ();
    }
}
catch (NumberFormatException error)
{
    _errorMessage.setText ("Enter a number");
    _finalRLabel = new JLabel ("Final r");
}
}
panel.add(_trueAnomalyTextField);
panel.add(_eorTextField);
panel.add(_trueAnomalyValueLabel = new JLabel ("True Anomaly in
degrees"));
panel.add(_trueAnomalyValueLabel);
}
trueAnomalyTextField = new JTextField(_textBoxLength);
trueAnomalyTextField.setEditable (true);
trueAnomalyTextField.setText ("0.0");
trueAnomalySet = true;
trueAnomalyTextField.addActionListener (new
ActionListener ()
{
    public void actionPerformed(ActionEvent e)
    {
        try
        {
            String testString =
trueAnomalyTextField.getText ();
            Double.valueOf (testString).doubleValue ();
            trueAnomalyValue =
trueAnomalyValue;
            while (_trueAnomalyValue < 0.0)
                trueAnomalyValue += 360.0;
            while (_trueAnomalyValue > 360.0)
                trueAnomalyValue -= 360.0;
            trueAnomalySet = true;
            errorMessage.setText ("");
            finalTextField.requestFocus ();
            checkOKStatus ();
        }
        catch (NumberFormatException error)
        {
            _errorMessage.setText ("Enter a number");
            _trueAnomalySet = false;
        }
    }
});
panel.add(_trueAnomalyTextField);
}
}

```

```

panel.add(_finalRLabel);

_finalRTextField = new JTextField(_textBoxLength);
_finalRTextField.setEditable (true);
_finalRSet = false;
_finalRTextField.addActionListener(new ActionListener()
{
    public void actionPerformed(ActionEvent e)
    {
        try
        {
            String testString =
_finalRTextField.getText();
Double.valueOf(testString).doubleValue();
_finalRSet = true;
ErrorMessage.setText("");
if (_selection !=2 && _finalRValue <
_radius[_body])
{
    finalRSet = false;
errorMessage.setText("Final r must be
> " + _radius[_body]/_factor);
else if (_selection == 2 && _finalRValue <
0.0)
{
    finalRSet = false;
errorMessage.setText("Final h must be
> 0");
}
else
    checkOKStatus ();
}
catch (NumberFormatException error)
{
    errorMessage.setText("Enter a number");
}
_finalRSet = false;
}
});
panel.add(_finalRTextField);

private JPanel getUnitPanel ()
{
    JPanel panel = new JPanel ();
    factor = 1.0;
    panel.setLayout (new BoxLayout(panel,
BoxLayout.Y_AXIS));
    AUButton = new JRadioButton ("AU", false);
    AUButton.addActionListener (new ActionListener()
{
    public void actionPerformed(ActionEvent event)
    {
        reset(1.4959965e11/factor);
        factor = 1.4959965e11;
        lengthUnitString = new String (" AU");
    }
});
ftButton = new JRadioButton ("feet", false);
ftButton.addActionListener (new ActionListener()
{
    public void actionPerformed(ActionEvent event)
    {
        reset(0.3048/factor);
        factor = 0.3048;
        lengthUnitString = new String (" ft");
    }
});
miButton = new JRadioButton ("miles", false);
miButton.addActionListener (new ActionListener()
{
    public void actionPerformed(ActionEvent event)
    {
        reset(1609.3/factor);
        factor = 1609.3;
        lengthUnitString = new String (" mi");
    }
});
nmButton = new JRadioButton ("nautical miles", false);
nmButton.addActionListener (new ActionListener()
{
    public void actionPerformed(ActionEvent event)
    {
        reset(1852.0/factor);
        factor = 1852.0;
        lengthUnitString = new String (" nm");
    }
});
public void actionPerformed(ActionEvent event)
{
    errorMessage = new JLabel ("");
    panel.add (_errorMessage);
    return panel;
}
}

```

```

        _eorTextField.postActionEvent();
        _trueAnomalyTextField.postActionEvent();
        _finalRTextField.postActionEvent();
        if (_aorSet && _eorSet && !_trueAnomalySet &&
            _finalRSet)
            finalizeOrbit();
        }
    }

    public void actionPerformed(ActionEvent event)
    {
        reset(1.0 / factor);
        factor = 1.0;
        _lengthUnitString = new String (" m");
    }
}

_kmButton = new JRadioButton ("kilometers", false);
_kmButton.addActionListener (new ActionListener()
{
    public void actionPerformed(ActionEvent event)
    {
        reset(1000.0 / factor);
        factor = 1000.0;
        _lengthUnitString = new String (" km");
    }
});
ButtonGroup group = new ButtonGroup();
group.add(_AUButton);
group.add(_ftButton);
group.add(_miButton);
group.add(_nmButton);
group.add(_mButton);
group.add(_kmButton);

panel.add(_AUButton);
panel.add(_ftButton);
panel.add(_nmButton);
panel.add(_mButton);
panel.add(_miButton);
panel.add(_kmButton);

JLabel temp = new JLabel ("");
panel.add(temp);

_defineOrbitOKButton = new JButton ("OK");
_defineOrbitOKButton.setEnabled (true);
ActionListener()
{
    public void actionPerformed(ActionEvent e)
    {
        _aorTextField.postActionEvent();
    }
}

```

```

        e = (ra - rp) / (ra + rp);
        initialOrbit = new Orbit (a, e,
Math.toRadians (_trueAnomalyValue), _body);
    }
    trajectory.setR2(_finalRValue);
    trajectory.setOrbit (_initialOrbit);
    bodyPath = trajectory.getBodyPath();
    _initialOrbitPath = trajectory.getInitialOrbitPath();
    finalOrbitPath = trajectory.getFinalOrbitPath();
    trajectory.initializePath();
    trajectoryPath = trajectory.getPath();
    satellite = trajectory.getSatellite();
    scale();
    updateOrbit();
    trajectoryPanel.repaint();
    statusLabel.setText("");
    orbitFrame.dispose();
}

private void reset(double value)
{
    if (_aorSet)
        aorValue *= value;
    if (_selection != 0 && _eorSet)
        eorValue *= value;
    if (_finalRSet)
        finalRValue *= _factor;
}

private void checkOKStatus ()
{
    if (_aorSet && _eorSet && _trueAnomalySet &&
_finalRSet)
    {
        defineOrbitOKButton.setEnabled (true);
        _defineOrbitOKButton.requestFocus();
    }
    return;
}

private void checkDefineVehicleOKStatus ()
{
    if (_massSet && _ispSet)
    {
        defineVehicleOKButton.requestFocus();
    }
    return;
}

private JPanel getBodyPanel ()
{
    JPanel panel = new JPanel ();
    panel.setLayout (new BoxLayout(panel,
BoxLayout.Y_AXIS));
    _body = 3;
    _bodyColor = Color.blue;
    sunButton = new JButton ("Sun");
    sunButton.addActionListener (new ActionListener()
{
    public void actionPerformed(ActionEvent event)
    {
        _body = 0;
        _bodyColor = Color.yellow;
        checkValues ();
    }
});
    mercuryButton = new JButton ("Mercury");
    mercuryButton.addActionListener (new ActionListener()
{
    public void actionPerformed(ActionEvent event)
    {
        _body = 1;
        _bodyColor = Color.gray;
        checkValues ();
    }
});
    venusButton = new JButton ("Venus");
    venusButton.addActionListener (new ActionListener()
{
    public void actionPerformed(ActionEvent event)
    {
        _body = 2;
        _bodyColor = Color.magenta;
        checkValues ();
    }
});
    earthButton = new JButton ("Earth");
    earthButton.addActionListener (new ActionListener()
{
    public void actionPerformed(ActionEvent event)
    {
        _body = 3;
        _bodyColor = Color.blue;
    }
});
}

```

```

checkValues () {
    {
        body = 8;
        bodyColor = Color.cyan;
        checkValues ();
    }
}

marsButton = new JRadioButton ("Mars", false);
marsButton.addActionListener (new ActionListener()
{
    public void actionPerformed(ActionEvent event)
    {
        body = 4;
        bodyColor = Color.red;
        checkValues ();
    }
});

jupiterButton = new JRadioButton ("Jupiter", false);
jupiterButton.addActionListener (new ActionListener()
{
    public void actionPerformed(ActionEvent event)
    {
        body = 5;
        bodyColor = Color.orange;
        checkValues ();
    }
});

saturnButton = new JRadioButton ("Saturn", false);
saturnButton.addActionListener (new ActionListener()
{
    public void actionPerformed(ActionEvent event)
    {
        body = 6;
        bodyColor = Color.cyan;
        checkValues ();
    }
});

uranusButton = new JRadioButton ("Uranus", false);
uranusButton.addActionListener (new ActionListener()
{
    public void actionPerformed(ActionEvent event)
    {
        body = 7;
        bodyColor = Color.cyan;
        checkValues ();
    }
});

neptuneButton = new JRadioButton ("Neptune", false);
neptuneButton.addActionListener (new ActionListener()
{
    public void actionPerformed(ActionEvent event)
    {
        body = 8;
        bodyColor = Color.cyan;
        checkValues ();
    }
});

PlutoButton = new JRadioButton ("Pluto", false);
PlutoButton.addActionListener (new ActionListener()
{
    public void actionPerformed(ActionEvent event)
    {
        body = 9;
        bodyColor = Color.lightGray;
        checkValues ();
    }
});

ButtonGroup group = new ButtonGroup ();
group.add (_sunButton);
group.add (_mercuryButton);
group.add (_venusButton);
group.add (_earthButton);
group.add (_marsButton);
group.add (_jupiterButton);
group.add (_saturnButton);
group.add (_uranusButton);
group.add (_neptuneButton);
group.add (_plutoButton);
panel.add (_earthButton);
panel.add (_marsButton);
panel.add (_jupiterButton);
panel.add (_mercuryButton);
panel.add (_venusButton);
panel.add (_neptuneButton);
panel.add (_plutoButton);
panel.add (_sunButton);
panel.add (_earthButton);
panel.add (_marsButton);
panel.add (_jupiterButton);
panel.add (_mercuryButton);
panel.add (_venusButton);
panel.add (_neptuneButton);
panel.add (_plutoButton);
return panel;
}

private void checkValues ()
{
    if (_aorSet)
    {
        if (_selection == 0)
        {
}
}
}

```

```

        if (_aorValue > 0.0 && _aorValue <
    }
    trajectory.setBody(_body);
    _initBodyPath = _trajectory.getBodyPath();

    private void restart ()
    {
        _trajectory.setMass(_initialMass);
        _trajectory.setOrbit(_initialOrbit);
        _trajectory.initializePath();
        _bodyPath = _trajectory.getBodyPath();
        _initialOrbitPath = _trajectory.getInitialOrbitPath();
        _finalOrbitPath = _trajectory.getFinalOrbitPath();
        _trajectoryPath = _trajectory.getPath();
        _satellite = _trajectory.getSatellite();
        _scale();
        _trajectoryPanel.repaint();
        _updateOrbit();
        _statusLabel.setText("");
    }

    public void setTrajectory (GeneralPath newPath)
    {
        _trajectoryPath = newPath;
        _scale();
    }

    private void newScale()
    {
        _bodyPath = (GeneralPath)_initBodyPath.clone();
        _initialOrbitPath =
        = (GeneralPath)_initInitialOrbitPath.clone();
        _finalOrbitPath =
        (GeneralPath)_initFinalOrbitPath.clone();
        _plotPath = (GeneralPath)_trajectoryPath.clone();
        _max = _initialMax* _scaleFactor;
        double scale = ((double)_dimension) / _max;
        transform = new AffineTransform();
        transform.setToIdentity();
        transform.scale(1.0, -1.0);
        _bodyPath.transform(_transform);
        _plotPath.transform(_transform);
        _initialOrbitPath.transform(_transform);
    }

    if (_finalRSet)
    {
        if (_finalRValue < _radius[_body])
        {
            finalRSet = false;
            finalRTextField.setText("");
            errorMessage.setText("Final r must be > " +
        }
        _radius[_body] / _factor;
        aorTextField.setText("a must be > " +
    }
    aorSet = false;
    aorTextField.setText("a must be > " +
    }
    aorTextField.requestFocus();
}

else if (_selection == 1)
{
    if (_aorValue < _radius[_body])
    {
        aorSet = false;
        errorMessage.setText("ra must be > " +
    }
    _radius[_body] / _factor;
    aorTextField.setText("");
    aorTextField.requestFocus();
}

if (_eorSet && _selection == 1)
{
    if (_eorValue < _radius[_body])
    {
        eorSet = false;
        errorMessage.setText("rp must be > " +
    }
    _radius[_body] / _factor;
    eorTextField.setText("");
    eorTextField.requestFocus();
}

if (_finalRSet)
{
    if (_finalRValue < _radius[_body])
    {
        finalRSet = false;
        finalRTextField.setText("");
        errorMessage.setText("Final r must be > " +
    }
    _radius[_body] / _factor;
}

```

```

    _finalOrbitPath.transform(_transform);
    _satellite.transform(_transform);
    transform.setToIdentity();
    transform.translate(-_centerX, -_centerY);
    bodyPath.transform(_transform);
    plotPath.transform(_transform);
    initialOrbitPath.transform(_transform);
    finalOrbitPath.transform(_transform);
    satellite.transform(_transform);

    transform.setToIdentity();
    transform.scale(_scale, _scale);
    plotPath.transform(_transform);
    bodyPath.transform(_transform);
    initialOrbitPath.transform(_transform);
    finalOrbitPath.transform(_transform);
    satellite.transform(_transform);

    transform.setToIdentity();
    transform.translate(_max*_scale*0.5, _max*_scale*0.5);
    plotPath.transform(_transform);
    bodyPath.transform(_transform);
    initialOrbitPath.transform(_transform);
    finalOrbitPath.transform(_transform);
    satellite.transform(_transform);

}

private void scale()
{
    GeneralPath testPath = new
    GeneralPath(_trajectoryPath);
    testPath.append(_bodyPath, false);
    testPath.append(_initialOrbitPath, false);
    testPath.append(_finalOrbitPath, false);
    initBodyPath = (GeneralPath)_bodyPath.clone();
    initInitialOrbitPath =
    (GeneralPath)_initialOrbitPath.clone();
    initFinalOrbitPath =
    (GeneralPath)_finalOrbitPath.clone();
    double width = testPath.getBounds2D().getWidth();
    double height = testPath.getBounds2D().getHeight();
    max = width;
    if (height > max)
        max = height;
    initialMax = max;
}

```

_centerX = testPath.getBounds2D().getCenterX();
 _centerY = testPath.getBounds2D().getCenterY();
 double scale = ((double)_dimension) / _max;

plotPath = _trajectoryPath;
 transform = new AffineTransform();
 transform.setToIdentity();
 transform.scale(1.0, -1.0);
 bodyPath.transform(_transform);
 plotPath.transform(_transform);
 initialOrbitPath.transform(_transform);
 finalOrbitPath.transform(_transform);
 satellite.transform(_transform);
 transform.setToIdentity();
 transform.translate(-_centerX, -_centerY);
 bodyPath.transform(_transform);
 plotPath.transform(_transform);
 initialOrbitPath.transform(_transform);
 finalOrbitPath.transform(_transform);
 satellite.transform(_transform);
 transform.setToIdentity();
 transform.translate(_centerX, -_centerY);
 bodyPath.transform(_transform);
 plotPath.transform(_transform);
 initialOrbitPath.transform(_transform);
 finalOrbitPath.transform(_transform);
 satellite.transform(_transform);
 transform.setToIdentity();
 transform.scale(_scale, _scale);
 plotPath.transform(_transform);
 bodyPath.transform(_transform);
 initialOrbitPath.transform(_transform);
 finalOrbitPath.transform(_transform);
 satellite.transform(_transform);
 transform.setToIdentity();
 transform.translate(_max*_scale*0.5, _max*_scale*0.5);
 plotPath.transform(_transform);
 bodyPath.transform(_transform);
 initialOrbitPath.transform(_transform);
 finalOrbitPath.transform(_transform);
 satellite.transform(_transform);
 transform.setToIdentity();
 transform.translate(_max*_scale*0.5, _max*_scale*0.5);
 plotPath.transform(_transform);
 bodyPath.transform(_transform);
 initialOrbitPath.transform(_transform);
 finalOrbitPath.transform(_transform);
 satellite.transform(_transform);
 private void rescale()
 {
 plotPath = (GeneralPath)_trajectoryPath.clone();
 transform = new AffineTransform();
 transform.setToIdentity();
 transform.scale(1.0, -1.0);
 }
}

```

    _plotPath.transform(_transform);
    _satellite.transform(_transform);

    transform.setToIdentity();
    -transform.translate(-_centerX, -_centerY);
    -plotPath.transform(_transform);
    -satellite.transform(_transform);

    double scale = ((double) dimension) / max;
    transform.setToIdentity();
    -transform.scale(scale, scale);
    -plotPath.transform(_transform);
    -satellite.transform(_transform);

    transform.setToIdentity();
    -transform.translate(max*scale*0.5, max*scale*0.5);
    -plotPath.transform(_transform);
    -satellite.transform(_transform);
}

public void actionPerformed(ActionEvent p1)
{
    errorStatus = trajectory.update();
    -trajectoryPath = trajectory.getPath();
    -satellite = trajectory.getSatellite();
    updateOrbit();
    drawOrbit();
    drawDiagram();
    if (_errorStatus == 0)
    {
        return;
    }
    else if (_errorStatus == 1)
    {
        pause();
        statusLabel.setText("Final distance Reached");
    }
    else if (_errorStatus == 2)
    {
        pause();
        statusLabel.setText("Impacted surface");
    }
    else if (_errorStatus == 3)
    {
        pause();
    }
}

StatusLabel.setText("Exhausted all mass");

    _statusLabel.setText("Exhausted all mass");
}

private void drawOrbit()
{
    rescale();
    -trajectoryPanel.repaint();
}

private void drawDiagram()
{
}

public void start()
{
    startAnimation();
}

public void stop()
{
    stopAnimation();
}

public synchronized void startAnimation()
{
    if (!paused)
    {
        if (!timer.isRunning())
            -timer.start();
    }
}

public synchronized void stopAnimation()
{
    if (_timer.isRunning())
    {
        -timer.stop();
    }
}

class PlotPanel extends JPanel
{
}

```



```

private double
private static final double [] _radius = {696000.0e3,
2487.0e3, 6187.0e3,
6378.0e3, 3380.0e3, 71370.0e3, 60400.0e3, 23530.0e3,
22320.0e3, 7016.0e3};
private final double [] _mu = {1.327e20, 2.168e13,
3.248e14, 3.986e14,
4.297e13, 1.267e17, 3.791e16, 5.788e15, 6.832e15, 9.965e11,
4.903e12};

private void initComponents () {
}

// Variables declaration - do not modify
// End of variables declaration
public Trajectory (int body, double R1, double R2,
ArrayList segment)
{
    _initialR = R1;
    _finalR = R2;
    _TOL = _tolFactor* _initialR;
    if (R2 < R1)
    {
        _direction = -1.0;
        _TOL = _tolFactor* _finalR;
    }
    body = body;
    currentInterpolation = 0;
    interpolationCount = segment.size();
    segment = segment;
    currentSegment =
(Interpolation)_segment.get(_currentInterpolation);
    position = _currentOrbit.getPosition ();
    velocity = _currentOrbit.getVelocity ();
    r = 0.05* _position.getMagnitude ();
    initializePath ();
    double r = _position.getMagnitude ();
    acceleration = (_position.unitVector () .times (
1.0* mu[_body]/(r*r));
}

public void setMass (double newMass)
{
    mass = newMass;
}

public void setIsp (double newIsp)
{
    Isp = newIsp;
}

(Interpolation)_segment.get(_currentInterpolation);
    currentOrbit = new Orbit (_initialR, 0.0, 0, _body);
    position = _currentOrbit getPosition ();
    velocity = _currentOrbit.getVelocity ();
    initializePath ();
    r = 0.05* _position.getMagnitude ();
    double r = _position.getMagnitude ();
    acceleration = (_position.unitVector () .times (
1.0* mu[_body]/(r*r));
}

public Trajectory (int body, Orbit initialOrbit, double R2,
ArrayList segment)
{
    currentOrbit = initialOrbit;
    finalR = R2;
    initialR = _currentOrbit.getRp ();
    TOL = _tolFactor* initialR;
    if (R2 < _initialR)
    {
        direction = -1.0;
        TOL = _tolFactor* _finalR;
    }
    body = body;
    currentInterpolation = 0;
    interpolationCount = segment.size();
    segment = segment;
    currentSegment =
(Interpolation)_segment.get(_currentInterpolation);
    position = _currentOrbit.getPosition ();
    velocity = _currentOrbit.getVelocity ();
    r = 0.05* _position.getMagnitude ();
    initializePath ();
    double r = _position.getMagnitude ();
    acceleration = (_position.unitVector () .times (
1.0* mu[_body]/(r*r));
}

public Orbit getOrbit ()
{
}

```

```

        return _currentOrbit;
    }

    public int getErrorStatus ()
    {
        return _errorStatus;
    }

    public double getTime()
    {
        return _time;
    }

    public MyVector getAcceleration()
    {
        return _acceleration;
    }

    public void setLimit (int newLimit)
    {
        _limit = newLimit;
    }

    public void setBody(int newBody)
    {
        _body = newBody;
    }

    public void setR2 (double newR2)
    {
        _finalR = newR2;
        if (_finalR < _currentOrbit.getRp ())
        {
            _direction = -1.0;
            _TOL = _tolFactor*_finalR;
        }
        else
            _TOL = _tolFactor*rp;
        return;
    }

    public void setSegment (ArrayList newSegment)
    {
        _segment = newSegment;
        _currentInterpolation = 0;
        _interpolationCount = _segment.size();
        _currentSegment =
        (Interpolation) _segment.get(_currentInterpolation);
    }

    public GeneralPath getPath ()
    {
        _path = _myPath.getPath();
        return _path;
    }

    public MyVector getThrustVector ()
    {
        return _thrustVector;
    }

    public void initializePath ()
    {
        _myPath = new MyGeneralPath();
        _Point2D datapoint = new Point2D.Double
        (_position.getX(), _position.getY());
        _myPath.moveTo(datapoint);
        _time = 0.0;
        _path = _myPath.getPath();
    }

    public GeneralPath getInitialOrbitPath ()
    {
        GeneralPath startOrbit = _currentOrbit.plotOrbit(1000);
        return startOrbit;
    }

    public GeneralPath getFinalOrbitPath ()

```

```

    {
        Ellipse2D endOrbit = new Ellipse2D.Double(-_finalR, -
        _finalR,
        2.0*_finalR);
        GeneralPath path = new GeneralPath(endOrbit);
        return path;
    }

    public GeneralPath getBodyPath ()
    {
        Ellipse2D planet = new Ellipse2D.Double(-
        _radius[_body], -_radius[_body],
        _radius[_body]*2.0,
        -_radius[_body]*2.0);
        GeneralPath path = new GeneralPath(planet);
        return path;
    }

    public GeneralPath getSatellite()
    {
        double x = _position.getX();
        double y = _position.getY();
        double x1 = x - _r;
        double x2 = x + _r;
        double y1 = y - _r;
        double y2 = y + _r;
        Ellipse2D circle = new Ellipse2D.Double(x1,y1,_r*2.0,
        -_r*2.0);
        GeneralPath satellite = new GeneralPath(circle);
        return satellite;
    }

    public int update ()
    {
        MyGeneralPath newPath = new
        MyGeneralPath(Shape)_myPath.getPath();
        int count = 0;
        Point2D dataPoint = new Point2D.Double (x, y);
        newPath.moveTo(dataPoint);
        double dist = _position.getMagitude();
        for (count = 0; count < _limit && dist* _direction <
        -_finalR* _direction && dist > _radius[_body] && _radius[_body]
        _mass > 0.0; count++)
        {
            doWork();
            dist = _position.getMagitude();
            x = _position.getX();
            y = _position.getY();
            dataPoint = new Point2D.Double (x, y);
            newPath.lineTo(dataPoint);
            errorStatus = 0;
            if (dist* _direction >= -finalR* _direction)
                errorStatus = 1;
            else if (dist <= -radius[_body])
                errorStatus = 2;
            else if (_mass <= 0.0)
                errorStatus = 3;
            _myPath.reset();
            _myPath = newPath;
            _path = (GeneralPath)_myPath.getPath().clone();
            return _errorStatus;
        }
    }

    private void doWork ()
    {
        getCurrentSegment ();
        _delta = calcDelta (_delta);
        _time += _delta;
        _mass -= _thrust/(_Isp*_g0)*_delta;
        _currentOrbit = new Orbit (_position, _velocity,
        _body);
    }

    private Interpolation getCurrentSegment ()
    {
        if (_currentSegment.withinBounds (_currentOrbit))
            return _currentSegment;
        else
        {
            do
            {
                currentInterpolation++;
                if (_currentInterpolation ==
                _interpolationCount)
                    _currentInterpolation = 0;
                _currentSegment =
                (Interpolation)_segment.get (_currentInterpolation);
            }
        }
    }
}

```

```

        } while
        {
            delta *= 0.5;
        }
    }

    private double calcDelta (double delta)
    {
        MyVector R1_2, V1_2, R2, V2, R1, A1, A2;
        MyVector gravityAcceleration, engineAcceleration,
        normal;
        double r, errorx, errory, max;
        Orbit newOrbit;
        final double mu = _mu[_body];
        do
        {
            r = _position.getMagnitude ();
            gravityAcceleration =
                position.unitVector().times(-1.0*mu/ (r*r));
            _currentSegment.getThrust (_currentOrbit).times(1.0/_mass);
            _currentSegment.engineAcceleration =
                thrustVector = engineAcceleration.times(_mass);
            _thrust = _thrustVector.getMagnitude();
            /* */
            * @author John E. Weglian
            * @version 2.0
            A1 = gravityAcceleration.plus(engineAcceleration);
            V1 = -velocity.plus(A1.times(delta));
            R1 = _position.plus(V1.times(delta));
            V1_2 = -velocity.plus(A1.times(0.5*delta));
            R1_2 = _position.plus(V1.times(0.5*delta));
            r = R1_2.getMagnitude();
            newOrbit = new Orbit (R1_2, V1_2, _body);
            gravityAcceleration = (R1_2.unitVector()).times(
                1.0*mu/ (r*r));
            engineAcceleration =
                _currentSegment.getThrust (newOrbit).times(1.0/_mass);
            A2 = gravityAcceleration.plus(engineAcceleration);
            V2 = V1_2.plus(A2.times(delta*0.5));
            R2 = R1_2.plus(V2.times(delta*0.5));
            errorx = Math.abs (R2.getX() - R1.getX());
            errory = Math.abs (R2.getY() - R1.getY());
            if (errorx > errory)
                max = errorx;
            else
                max = errory;
            if (max > _TOL)
                delta *= 1.2;
            else if (max < _TOL*0.5)
                delta *= 0.5;
        } while (max > _TOL);
        position = R1;
        velocity = V1;
        acceleration = A1;
        return delta;
    }
}

/*
 * Orbit.java
 *
 * Created on May 16, 2001, 3:16 PM
 */
public class Orbit extends java.lang.Object
{
    private double h;
    private double phi;
    private double a;
    private double p;
    private double e;
    private double rp;
    private double ra;
    private double vp;
    private double va;
    private double period;
    private double trueAnomaly;
}

```

```

private double _u;
private double _M;
private double _time;
private double _rotation;

private int _body;

private MyVector _position;
private MyVector _velocity;

private final double [] mu = {1.327e20, 2.168e13,
3.248e14, 3.986e14, 4.297e13, 1.267e17, 3.791e16,
5.788e15, 6.832e15, 9.965e11, 4.903e12};

/** Creates new Orbit */
public Orbit()
{
}

public Orbit (MyVector position, MyVector velocity, int
body)
{
    body = body;
    double r = position.getMagnitude ();
    double v = velocity.getMagnitude ();
    _position = position;
    _velocity = velocity;
    _energy = v*v*0.5 - mu [body]/r;
    _h = _position.crossMagnitude (_velocity);
    double temp = _h/ (r*v);
    if (temp > 1.0)
        _phi = 0.0;
    else if (temp < -1.0)
        _phi = Math.acos (_h/ (r*v));
    else
        if (_position.dot (_velocity) < 0.0)
            _phi *= -1.0;
        _a = -1.0* mu [body]* 0.5/ _energy;
        _p = _h* _h/ mu [body];
        if (_p/ _a > 1.0)
            _e = 0.0;
        else
            _e = Math.sqrt (1.0 - _p/ _a);
    _rp = _a* (1.0 - _e);
    _ra = _a* (1.0 + _e);
    _vp = _h/ _rp;
}

public void calcTime()
{
    double temp = 1.0/ e* (_p/r - 1.0);
    if (temp > 1.0)
        _trueAnomaly = 0.0;
    else if (temp < -1.0)
        _trueAnomaly = Math.PI;
    else
        _trueAnomaly = Math.acos (temp);
    if (_position.dot (_velocity) < 0.0)
        _trueAnomaly *= -1.0;
}

public void rotation (double rotation)
{
    _rotation = _position.getTheta () - _trueAnomaly;
}

public void calcTime()
{
    double temp = 1.0/ e* (_p/r - 1.0);
    if (temp > 1.0)
        _trueAnomaly = 0.0;
    else if (temp < -1.0)
        _trueAnomaly = Math.PI;
    else
        _trueAnomaly = Math.acos (temp);
    if (_position.dot (_velocity) < 0.0)
        _trueAnomaly *= -1.0;
}

public void calcTime (double rotation)
{
    _rotation = _position.getTheta () - _trueAnomaly;
}

public void calcTime (double rotation, double phi)
{
    _rotation = rotation;
    _phi = phi;
}

public void calcTime (double rotation, double phi, double angle)
{
    _rotation = rotation;
    _phi = phi;
    _angle = angle;
}

public void calcTime (double rotation, double phi, double angle, double
cw)
{
    _rotation = rotation;
    _phi = phi;
    _angle = angle;
    if (cw)
        _angle = -Math.PI + _phi;
    else
        _angle = Math.PI - _phi;
}

public void calcTime (double rotation, double phi, double angle, double
cw, int body)
{
    _rotation = rotation;
    _phi = phi;
    _body = body;
    _angle = angle;
    if (cw)
        _angle = -Math.PI + _phi;
    else
        _angle = Math.PI - _phi;
    _velocity.setUnitVector ().rotate (angle);
    _velocity.setMagnitude (r);
    _energy = v*v*0.5 - mu [_body]/r;
    _h = _position.crossMagnitude (_velocity);
    _a = -1.0* mu [body]* 0.5/ _energy;
    _p = _h* _h/ mu [body];
    if (_p/ _a > 1.0)
        _e = 0.0;
    else
        _e = Math.sqrt (1.0 - _p/ _a);
    _rp = _a* (1.0 - _e);
    _ra = _a* (1.0 + _e);
    _vp = _h/ _rp;
}

```

```

    _va = -h/_ra;
    double temp = (_p/r - 1.0) / _e;
    if (temp > 1.0)
        _trueAnomaly = 0.0;
    else if (temp < -1.0)
        _trueAnomaly = Math.PI;
    else
    {
        trueAnomaly = Math.acos(temp);
        if (_position.dot(_velocity) < 0.0)
            _trueAnomaly *= -1.0;
    }
    rotation = _position.getTheta() - _trueAnomaly;
    calcTime();
}

public Orbit (double a, double e, double trueAnomaly, int
body, double rotation)
{
    rotation = rotation;
    _a = a;
    _e = e;
    _trueAnomaly = trueAnomaly;
    body = body;
    energy = -mu[_body]*0.5/_a;
    p = a*(1.0 - e*e);
    h = Math.sqrt(p*_mu[_body]);
    rp = a*(1.0 - e);
    ra = a*(1.0 + e);
    vp = h/rp;
    va = h/_ra;
    setVectors();
    calcTime();
}

public Orbit (double a, double e, double trueAnomaly, int
body)
{
    this(a, e, trueAnomaly, body, 0.0);
}

public double getEnergy ()
{
    return _energy;
}

public double getH ()
{
    if (_e < 1.0)

```

```

        return _period;
    else
        return -1.0;
}

public double getTrueAnomaly ()
{
    double answer = Math.toDegrees(_M);
    if (answer < 0.0)
        answer += 360.0;
    return answer;
}

public double getMeanAnomaly ()
{
    double answer = Math.toDegrees(_M);
    if (answer < 0.0)
        answer += 360.0;
    return answer;
}

public MyVector getPosition ()
{
    return _position;
}

public MyVector getVelocity ()
{
    return _velocity;
}

public double getTime ()
{
    return _time;
}

public double getRotation ()
{
    return _rotation;
}

private void solveEllipse ()
{
    double min, mid, max, test, eval;
    double r, v;
    double newTime = _time;
    double TOL = 0.00001;
    while (newTime < 0.0)
        newTime += _period;
    while (newTime > _period)
        newTime -= _period;
    _M = newTime*2.0*Math.PI/_period;
    if (_M < Math.PI)
    {
        min = 0.0;
        max = Math.PI;
    }
    else
    {
        min = Math.PI;
        max = 2.0*Math.PI;
    }
    do
    {
        mid = (min + max)*0.5;
        test = mid - _e*Math.sin(mid);
        if (test < _M)
            min = mid;
        else
            max = mid;
        eval = Math.abs(test - _M);
    } while (eval > TOL);
    _u = mid;
    double check = Math.cos(_u);
    if (_M < Math.PI)
    {
        min = 0.0;
        max = Math.PI;
    }
    else
    {
        min = -Math.PI;
        max = 0.0;
    }
    do
    {
        mid = (min + max)*0.5;
        test = (_e + Math.cos(mid)) / (1.0 +
_e*Math.cos(mid));
        if (_u < Math.PI && test > check) || (_u > Math.PI
&& test < check))
            min = mid;
        else
    }
}

```

```

max = mid;
eval = Math.abs(test - check);
} while (eval > TOL);
trueAnomaly = mid;
}

private void solveHyperbola ()
{
    double min, mid, max, test, eval;
    double r, v;
    double TOL = 0.000001;
    M = t * Math.sqrt(-1.0 * a * a * mu[_body]);
    if (_M == 0.0)
    {
        u = 0.0;
        trueAnomaly = 0.0;
        return;
    }
    double MTOL = M*TOL;
    if (_time > 0.0)
    {
        min = 0.0;
        max = 10.0;
        while ((e*MyMath.sinh(max) - max) < M)
            max *= 10.0;
    }
    else
    {
        max = 0.0;
        min = -10.0;
        while ((e*MyMath.sinh(min) - min) > M)
            min *= 10.0;
    }
    do
    {
        mid = (min + max)*0.5;
        test = e*MyMath.sinh(mid) - mid;
        if (_M > 0.0 && test > M) || (_M < 0.0 && test < -M)
            max = mid;
        else
            min = mid;
        eval = Math.abs(test - M);
        } while (eval > MTOL);
        u = mid;
    }

    double check = MyMath.cosh(_u);
    double UTOL = TOL*check;
    if (_M > 0.0)
    {
        min = 0.0;
        max = Math.acos(-1.0/_e);
    }
    else
    {
        min = -Math.acos(-1.0/_e);
        max = 0.0;
    }
    do
    {
        mid = (min + max)*0.5;
        test = (e + Math.cos(mid))/(1.0 +
        e*Math.cos(mid));
        if ((test > check && M > 0.0) || (test < check &&
_M < 0.0))
            max = mid;
        else
            min = mid;
        eval = Math.abs(test - check);
        } while (eval > UTOL && Math.abs(min-max) > TOL);
        _trueAnomaly = mid;
    }

    private void solveParabola()
    {
        double min, mid, max, test, eval;
        double TOL = 0.000001;
        if (_time < 0.0)
        {
            min = -Math.PI;
            max = 0.0;
        }
        else
        {
            min = 0.0;
            max = Math.PI;
        }
        do
        {
            mid = (min + max)*0.5;
            test = e*MyMath.sinh(mid) - mid;
            if (_M > 0.0 && test > M) || (_M < 0.0 && test < -M)
                max = mid;
            else
                min = mid;
            eval = Math.abs(test - M);
            } while (eval > MTOL);
            u = mid;
        }

        mid = (min + max)*0.5;
        test = h * h * 0.5 / (_mu[_body] *
_mu[_body]) * (Math.tan(mid*0.5) +
Math.pow(Math.tan(mid*0.5), 3.0)/3.0);
    }
}

```

```

    if (test > _time && _time > 0.0) || (test < _time
&& _time < 0.0)
        M = e*MyMath.sinh(_u) - _u;
        _time = Math.sqrt(-1.0*_a*_a*_a/_mu[_body])*_M;
    }
    else
    {
        min = mid;
        eval = Math.abs(test - _time);
        _trueAnomaly = mid;
    }
}

private void setVectors ()
{
    double r = -p/(1.0 + e*Math.cos(_trueAnomaly));
    double v = Math.sqrt(2.0*(_energy + _mu[_body]/r));
    double temp = h/(r*v);
    if (temp > 1.0)
        phi = 0.0;
    else if (temp < -1.0)
        phi = Math.PI;
    else
        phi = Math.acos(h/(r*v));
    if (_trueAnomaly < 0.0 || _trueAnomaly > Math.PI)
        phi *= -1.0;
    position = new MyVector (r, (_trueAnomaly +
    -rotation), true);
    double theta = _trueAnomaly + Math.PI*0.5 - phi +
    rotation;
    velocity = new MyVector (v, theta, true);
}
private void calcTime ()
{
    if (e < 1.0)
    {
        period =
2.0*Math.PI/Math.sqrt(_mu[_body])*Math.pow(_a, 1.5);
        u = Math.acos(_e + Math.cos(_trueAnomaly)) / (1.0 +
e*Math.cos(_trueAnomaly));
        if (_trueAnomaly < 0.0 || _trueAnomaly > Math.PI)
            u = -u;
        M = -u - e*Math.sin(u);
        _time = -M*period*0.5/Math.PI;
    }
    else if (_e > 1.0)
    {
        u = MyMath.acosh ((_e +
Math.cos(_trueAnomaly)) / (1.0 + e*Math.cos(_trueAnomaly)));
        time = Math.sqrt(_mu[_body]*_mu[_body]) * (Math.tan(_trueAnomaly)*0.5
+ Math.pow(Math.tan(_trueAnomaly*0.5), 3.0) / 3.0);
    }
}
public Orbit setTime(double time)
{
    time = time;
    if (_e < 1.0)
        solveEllipse();
    else if (_e > 1.0) // Hyperbolic
        solveHyperbola();
    else
        solveParabola();
    setVectors();
    return this;
}
public Orbit addTime(double moreTime)
{
    time += moreTime;
    if (_e < 1.0)
        solveEllipse();
    else if (_e > 1.0) // Hyperbolic
        solveHyperbola();
    else
        solveParabola();
    setVectors();
    return this;
}
public Orbit toDistance (double distance)
{
    if (distance > _ra && _e < 1.0)
    {
        System.out.println("Error. Distance > ra");
        System.out.println("ra = " + _ra);
        return null;
    }
    else if (distance < _rp)

```

```

{
    System.out.println("Error. Distance < rp");
    System.out.println("rp = " + rp);
    return null;
}
else if (_e == 0.0)
{
    System.out.println("Error. Not defined for
circular orbit");
    return null;
}
double r = distance;
double newTrueAnomaly;
boolean flip = false;
if ((_p/r - 1.0) / _e > 1.0)
    newTrueAnomaly = 0.0;
else
    newTrueAnomaly = Math.acos(((_p/r - 1.0) / _e);
if (-newTrueAnomaly > trueAnomaly)
    newTrueAnomaly *= -1.0;
else if (-trueAnomaly > newTrueAnomaly && _e < 1.0)
    flip = true;
trueAnomaly = newTrueAnomaly;
setVectors();
calcTime();
if (flip)
    time += _period;
return this;
}

public Orbit setTrueAnomaly (double newAnomaly)
{
    trueAnomaly = newAnomaly;
    setVectors();
    calcTime();
    return this;
}

public GeneralPath plotOrbit (int points)
{
    double maxDist = 100.0*_rp;
    return plotOrbit(points, maxDist);
}
public GeneralPath plotOrbit (int points, GeneralPath path)
{
    double maxDist = 100.0*_rp;
    return plotOrbit(points, maxDist, path);
}
public GeneralPath plotOrbit (int points, double maxDist)
{
    GeneralPath path = new GeneralPath();
    return plotOrbit(points, maxDist, path);
}
public GeneralPath plotOrbit (int points, double maxDist,
GeneralPath path)
{
    double oldTrueAnomaly = _trueAnomaly;
    double DELTA, max, min;
    MyGeneralPath myPath = new MyGeneralPath();
    Point2D dataPoint = new Point2D.Double(0.0, 0.0);
    myPath.moveTo (dataPoint);
    myPath.lineTo (dataPoint);
    if (_e < 1.0)
    {
        min = -Math.PI;
        max = Math.PI;
    }
    else
    {
        toDistance(maxDist);
        if (_trueAnomaly > 0.0)
            min = _trueAnomaly *= -1.0;
        else
            max = -min;
    }
    DELTA = (max - min) / ((double) (points));
    _trueAnomaly = min;
    setVectors();
    dataPoint = new Point2D.Double (_position.getX(),
_position.getY());
    myPath.moveTo(dataPoint);
    myPath.lineTo(dataPoint);

    for (_trueAnomaly = min; _trueAnomaly <= max;
_trueAnomaly += DELTA)
    {
        setVectors();
        dataPoint = new Point2D.Double (_position.getX(),
_position.getY());
        myPath.lineTo(dataPoint);
    }
    _trueAnomaly = oldTrueAnomaly;
}

```

```

private double
setVectors()
{
    calcTime();
    path.append((Shape) (myPath.getPath()), false);
    return path;
}

public Orbit deltaV (double velocityChange)
{
    double v = _velocity.getMagnitude();
    System.out.println("v = " + v);
    v += velocityChange;
    System.out.println("v now = " + v);
    Myvector newVelocity = _velocity.uniteVector();
    newVelocity.setMagnitude(v);
    Orbit result = new Orbit (_position, newVelocity,
    _body);
    return result;
}

public Orbit deltaV (MyVector velocityChange)
{
    MyVector newVelocity = _velocity.plus(velocityChange);
    Orbit result = new Orbit (_position, newVelocity,
    _body);
    return result;
}

/*
 * MyVector.java
 * Created on April 27, 2001, 12:42 PM
 */
/*
 * @author John E. Weglian
 * @version 1.0
 */

import java.lang.Math;

public class MyVector
{
    private double
    private double
    private double
}

```

```

    return _x;
}

public double getY ()
{
    return _y;
}

public double getTheta ()
{
    return _theta;
}

public double getDelta (MyVector b)
{
    return b.getTheta() - _theta;
}

public MyVector unitVector ()
{
    double newX = Math.cos (_theta);
    double newY = Math.sin (_theta);
    MyVector newVector = new MyVector (newX, newY);
    return newVector;
}

public void setMagnitude (double magnitude)
{
    _x = magnitude*Math.cos (_theta);
    _y = magnitude*Math.sin (_theta);
    _magnitude = magnitude;
}

public void setTheta (double theta)
{
    _theta = theta;
    _x = _magnitude*Math.cos (_theta);
    _y = _magnitude*Math.sin(_theta);
}

public MyVector rotate (double delta)
{
    setTheta (_theta + delta);
    return this;
}

public MyVector plus (MyVector addVector)
{
    double newX = _x + addVector.getX();
    double newY = _y + addVector.getY();
    MyVector newVector = new MyVector (newX, newY);
    return newVector;
}

public MyVector times (double mult)
{
    double newX = _x*mult;
    double newY = _y*mult;
    MyVector newVector = new MyVector (newX, newY);
    return newVector;
}

public double crossMagnitude (MyVector b)
{
    double delta = getDelta(b);
    return Math.abs (_magnitude*b.getMagnitude() *Math.sin(delta));
}

public double dot (Myvector b)
{
    double delta = getDelta (b);
    return _magnitude*b.getMagnitude() *Math.cos(delta);
}

}
/*
 * MyGeneralPath.java
 *
 * @author John E. Weglian
 * @version 1.0
 */
import java.awt.geom.GeneralPath;
import java.awt.geom.Point2D;
import java.awt.Shape;

public class MyGeneralPath
{

```

```

/*
 *  * Interpolation.java
 *  *  * Created on August 31, 2001, 7:00 PM
 */
public MyGeneralPath()
{
    _path = new GeneralPath();
}

public MyGeneralPath (Shape s)
{
    _path = new GeneralPath (s);
}

import java.lang.Math;

public GeneralPath lineTo (Point2D newPoint)
{
    float x = (float)newPoint.getX();
    float y = (float)newPoint.getY();
    _path.lineTo (x, y);
    return _path;
}

public GeneralPath moveTo (Point2D newPoint)
{
    float x = (float)newPoint.getX();
    float y = (float)newPoint.getY();
    _path.moveTo (x, y);
    return _path;
}

public GeneralPath getPath ()
{
    return _path;
}

public GeneralPath append(Shape s, boolean connect)
{
    _path.append (s, connect);
    return _path;
}

public void reset ()
{
    _path.reset();
}

/*
 *  * Interpolation.java
 *  *  * @author John E. Weglian
 *  *  * @version 1.0
 */
import java.lang.Object;

public class Interpolation extends java.lang.Object
{
    private boolean _useTrueAnomaly =
    private boolean _velocityBased =
    private double _startAngle;
    private double _endAngle;
    private double _startMagnitude;
    private double _endMagnitude;
    private double _angleConstant;
    private double _magnitudeConstant;

    /**
     * Creates new Interpolation */
    public Interpolation(boolean useTrueAnomaly, boolean
    velocityBased, double startAnomaly, double endAnomaly,
    double startAngle, double endAngle, double endAngle,
    double startMagnitude, double endMagnitude)
    {
        _useTrueAnomaly = useTrueAnomaly;
        _velocityBased = velocityBased;
        _startAnomaly = startAnomaly;
        _endAnomaly = endAnomaly;
        _startAngle = startAngle;
        _endAngle = endAngle;
        _startMagnitude = startMagnitude;
        _endMagnitude = endMagnitude;
    }
}

```

```

        - _startAnomaly);
        - _angleConstant = (_endAngle - _startAngle) / (_endAnomaly)
    }

    public double getStartAnomaly ()
    {
        return _startAnomaly;
    }

    public double getAngle (Orbit currentOrbit)
    {
        double x, y;
        if (_useTrueAnomaly)
            x = currentOrbit.getTrueAnomaly ();
        else
            x = currentOrbit.getMeanAnomaly ();
        y = (x - _startAnomaly) * _angleConstant + _startAngle;
        return y;
    }

    public double getMagnitude (Orbit currentOrbit)
    {
        double x, y;
        if (_useTrueAnomaly)
            x = currentOrbit.getTrueAnomaly ();
        else
            x = currentOrbit.getMeanAnomaly ();
        y = (x - _startAnomaly) * _magnitudeConstant +
            _startMagnitude;
        return y;
    }

    public MyVector getThrust (Orbit currentOrbit)
    {
        double angle, magnitude;
        angle = Math.toRadians (getAngle (currentOrbit));
        if (_velocityBased)
            angle+=currentOrbit.getVelocity ().getTheta ();
        else
            angle+=currentOrbit.getPosition ().getTheta ();
        magnitude = getMagnitude (currentOrbit);
        MyVector thrust = new MyVector (magnitude, angle,
            true);
        return thrust;
    }

    public double getEndAnomaly ()
    {
        return _endAnomaly;
    }

}

```

```

/*
 * @author John E. Weglian
 * @version 1.0
 */

```

```

import java.awt.geom.AffineTransform;
import java.awt.geom.Point2D;
import java.awt.geom.GeneralPath;
import java.awt.Graphics;
import java.awt.Graphics2D;
import java.awt.Color;
import javax.swing.JFrame;
import javax.swing.JPanel;
import java.awt.event.WindowAdapter;
import java.awt.event.WindowEvent;
import java.awt.BasicStroke;

```

```

public class MyPlot

```

```

    {
        private AffineTransform
        _transform
        = null;
        private GeneralPath
        _path
        = null;
        private int
        = 500;
        /**
         * Creates new MyPlot */
        public MyPlot(GeneralPath path)
        {
            _path = path;
            scale ();
        }
        public void setPath(GeneralPath newPath)
        {
            _path = newPath;
        }
        public void scale()
        {
            transform = new AffineTransform();
            transform.setToIdentity();
            double x = _path.getBounds2D().getCenterX();
            double y = _path.getBounds2D().getCenterY();
            transform.translate(-x, -y);
            transform.transform(_transform);
            double width = _path.getBounds2D().getWidth();
            double height = _path.getBounds2D().getHeight();
            double max = width;
            if (height > max)
                max = height;
            double scale = ((double) dimension) / max;
            transform.setToIdentity();
            transform.scale(scale, scale);
            path.transform(_transform);
            transform.setToIdentity();
            transform.translate(max * scale * 0.5, max * scale * 0.5);
            path.transform(_transform);
        }
        public void plot(String title)
        {
            JFrame frame = new JFrame(title);
            PlotPanel panel = new PlotPanel();
            frame.getContentPane().add(panel);
            frame.setSize(_dimension + 40, _dimension + 60);
            frame.show();
        }
    }
}

class PlotPanel extends JPanel
{
    public PlotPanel()
    {
        super();
    }
    public void paintComponent(Graphics g)
    {
        super.paintComponent(g);
        Graphics2D g2 = (Graphics2D) g;
        g2.setPaint(Color.black);
        g2.draw(_path);
    }
}
// class PlotPanel

/*
 * class MyPlot
 */
/*
 * SetUp.java
 *
 * Created on September 20, 2001, 3:26 PM
 */
/*
 * @author John E. Weglian
 * @version 1.0
 */
import javax.swing.JFrame;
import javax.swing.JButton;
import javax.swing.JLabel;
import javax.swing.JComboBox;
import javax.swing.JRadioButton;
import javax.swing.ButtonGroup;
import javax.swing.JTextField;
import javax.swing.BoxLayout;
import java.awt.event.ActionListener;
import java.awt.event.ActionEvent;
import java.text.NumberFormat;

```



```

private boolean _initAngleSet;
private boolean _endAngleSet;
private boolean _initThrustSet;
private boolean _endThrustSet;
private boolean _velocityBased;
private boolean _segmentsSet;
private double [] _scale;
private ArrayList<Segment> _segmentArrayList;
private Interpolation [] _segment;
private Trajectory _trajectory;
private String _errorMessage;
private String ("");
/** Creates new Interface */
public Setup()
{
    // 
    setUpSegments();
}

public void setUpSegments (Trajectory currentTrajectory,
double initialAnomaly)
{
    _trajectory = currentTrajectory;
    _initialAnomaly = initialAnomaly;
    segmentFrame = new JFrame ("Define trajectory
segments");
 JPanel anomalyPanel = getAnglePanel();
 JPanel anglePanel = getAnglePanel ();
 JPanel thrustPanel = getThrustPanel ();
 Container contentPane = segmentFrame.getContentPane();
 contentPane.setLayout( new GridLayout(1,3));
 contentPane.add (anomalyPanel);
}

private void startAnglePanel()
{
    contentPane.add (anglePanel);
    contentPane.add (thrustPanel);
    segmentFrame.pack ();
    segmentFrame.show ();
}

private JPanel getAnglePanel ()
{
    JPanel anomalyPanel = new JPanel ();
    anomalyPanel.setLayout (new BoxLayout (anomalyPanel,
BoxLayout.Y_AXIS));
    _anomalyLabel = new JLabel ("True Anomaly Definition");
    anomalyPanel.add (_anomalyLabel);
    _trueAnomalyButton = new JButton ("True Anomaly");
    _trueAnomalyButton.addActionListener (new ActionListener()
    {
        public void actionPerformed(ActionEvent event)
        {
            useTrueAnomaly = true;
            _anomalyLabel.setText ("True Anomaly
Definition");
        }
    });
    _meanAnomalyButton = new JButton ("Mean Anomaly",
true);
    _meanAnomalyButton.addActionListener (new ActionListener()
    {
        public void actionPerformed(ActionEvent event)
        {
            useTrueAnomaly = false;
            _anomalyLabel.setText ("Mean Anomaly
Definition");
        }
    });
    _trueAnomalyButton.setEnabled (! initAnomalySet);
    _meanAnomalyButton.setEnabled (! initAnomalySet);
    ButtonGroup group = new ButtonGroup ();
    group.add (_trueAnomalyButton);
    group.add (_meanAnomalyButton);
    anomalyPanel.add (_trueAnomalyButton);
    anomalyPanel.add (_meanAnomalyButton);
}

private void startThrustPanel()
{
    contentPane.add (thrustPanel);
}

private void startVelocityPanel()
{
    contentPane.add (velocityPanel);
}

```

```

anomalyPanel.add(_startAnomalyLabel);

String value = new String("");
value = Double.toString(_initialAnomaly);
_startAnomalyTextField = new JTextField(_textBoxLength);
_startAnomalyTextField.setText(value);
_startAnomalyTextField.setEditable (false);
_initAnomalySet = true;
_startAnomalyTextField.addActionListener(new
ActionListener()
{
    public void actionPerformed(ActionEvent e)
    {
        try
        {
            String testString =
                _startAnomalyTextField.getText();
            initialAnomaly =
Double.valueOf(testString).doubleValue();
            if (_endAnomaly < _initialAnomaly)
                endAnomaly += 360.0;
            if (_endAnomaly <= _initialAnomaly + 360.0)
            {
                anomalyErrorMessage.setText ("");
                _endAnomalySet = true;
            }
            else
            {
                anomalyErrorMessage.setText ("Beginning
and end anomaly must be within 360 degrees");
                _endAnomalySet = false;
            }
        }
        catch (NumberFormatException error)
        {
            anomalyErrorMessage.setText ("Enter a
number");
            _endAnomalySet = false;
        }
    }
}
);
_anomalyPanel.add(_startAnomalyTextField);

_endAnomalyLabel = new JLabel ("Ending Anomaly value in
degrees");
_anomalyPanel.add(_endAnomalyLabel);

_endAnomalyTextField = new JTextField(_textBoxLength);
_endAnomalyTextField.setEditable (true);
_endAnomalyTextField.requestFocus();
_endAnomalyTextField.addActionListener(new
ActionListener()
{
    public void actionPerformed(ActionEvent e)
    {
        try
        {
            String testString =
                _endAnomalyTextField.getText();
            endAnomaly =
Double.valueOf(testString).doubleValue();
            if (_endAnomaly < _initialAnomaly)
                endAnomaly += 360.0;
            if (_endAnomaly <= _initialAnomaly + 360.0)
            {
                anomalyErrorMessage.setText ("");
                _endAnomalySet = true;
            }
            else
            {
                anomalyErrorMessage.setText ("Beginning
and end anomaly must be within 360 degrees");
                _endAnomalySet = false;
            }
        }
        catch (NumberFormatException error)
        {
            anomalyErrorMessage.setText ("Enter a
number");
            _endAnomalySet = false;
        }
    }
}
);
_anomalyPanel.add(_endAnomalyTextField);

private JPanel getAnglePanel ()
{
    JPanel anglePanel = new JPanel ();
    anglePanel.setLayout (new BoxLayout(anglePanel,
BoxLayout.Y_AXIS));
    JLabel angleLabel = new JLabel ("Thrust Angle
Definition");
    anglePanel.add(angleLabel);
    JButton angleButton = new JButton ("Relative to
Velocity Vector", true);
    anglePanel.add(angleButton);
    JButton velocityAngleButton = new JButton ("Relative to
Velocity Angle", true);
    anglePanel.add(velocityAngleButton);
    velocityAngleButton.addActionListener (new
ActionListener()
{
    public void actionPerformed(ActionEvent e)
    {
        try
        {
            String testString =
                _velocityAngleTextField.getText();
            velocityAngle =
Double.valueOf(testString).doubleValue();
            if (_velocityAngle < _initialVelocityAngle)
                velocityAngle += 360.0;
            if (_velocityAngle <= _initialVelocityAngle + 360.0)
            {
                velocityErrorMessage.setText ("");
                _velocityAngleSet = true;
            }
            else
            {
                velocityErrorMessage.setText ("Beginning
and end velocity angle must be within 360 degrees");
                _velocityAngleSet = false;
            }
        }
        catch (NumberFormatException error)
        {
            velocityErrorMessage.setText ("Enter a
number");
            _velocityAngleSet = false;
        }
    }
}
);
_anomalyPanel.add(_velocityAngleTextField);

return anomalyPanel;
}
}
);

```



```

public void actionPerformed (ActionEvent e)
{
    JPanel thrustPanel = new JPanel ();
    thrustPanel.setLayout (new BoxLayout (thrustPanel,
    BoxLayout.Y_AXIS));
    JLabel thrustLabel = new JLabel ("Thrust Magnitude
Definition");
    thrustPanel.add (thrustLabel);
    _nUnitButton = new JButton ("N", true);
    _nUnitButton.addActionListener (new ActionListener()
    {
        public void actionPerformed (ActionEvent event)
        {
            if (_initThrustSet)
            {
                if (_initAngleSet && _initAnomalySet && _endAngleSet &&
                _initThrustSet && _endThrustSet)
                    defineSegment ();
                else
                {
                    errorMessage = "";
                    if (!_initAnomalySet)
                        errorMessage = "Need Initial Anomaly
";
                    if (!_endAngleSet)
                        errorMessage += "Need Final Anomaly";
                    anomalyErrorMessage.setText (_errorMessage);
                    errorMessage = "";
                    if (!_initAngleSet)
                        errorMessage = "Need Initial Angle ";
                    if (!_endAngleSet)
                        errorMessage += "Need Final Angle";
                    errorMessage += "Need Final Angle";
                    angleErrorMessage.setText (_errorMessage);
                    errorMessage = "";
                    if (!_initThrustSet)
                        errorMessage = "Need Initial Thrust
";
                    if (!_endThrustSet)
                        errorMessage += "Need Final Thrust";
                    thrustErrorMessage.setText (_errorMessage);
                }
            }
        }
    });
    angleErrorMessage = new JLabel ("");
    anglePanel.add (_angleErrorMessage);
    anglePanel.add (_OKButton);
    return anglePanel;
} //getAnomalyPanel

private JPanel getThrustPanel ()
{
    _angleErrorMessage = new JLabel ("");
    _anglePanel.add (_angleErrorMessage);
    _lbUnitButton = new JButton ("lb", true);
    _lbUnitButton.addActionListener (new ActionListener()
    {
        public void actionPerformed (ActionEvent event)
        {
            if (_initThrustSet)
            {
                _initialThrust *= 4.44822/_factor;

```

```

        }
        if (_endThrustSet)
        {
            _endThrustLabel = new JLabel ("Ending thrust value");
            _trustPanel.add(_endThrustLabel);
            _endThrustTextField = new JTextField(_textBoxLength);
            _endThrustTextField.setEditable (true);
            _endThrustTextField.addActionListener(new
                ActionListener()
            {
                public void actionPerformed(ActionEvent e)
                {
                    try
                    {
                        String testString =
                            _endThrustTextField.getText();
                        endThrust = factor *
                            Double.valueOf(testString).doubleValue();
                        endThrustSet = true;
                        _OKButton.requestFocus();
                    }
                    catch (NumberFormatException error)
                    {
                        thrustErrorMessage.setText ("Enter a
                            number");
                        endThrustSet = false;
                    }
                }
            });
            thrustPanel.add(_startThrustLabel);
            _startThrustTextField = new JTextField(_textBoxLength);
            _startThrustTextField.setEditable (true);
            _startThrustTextField.addActionListener(new
                ActionListener()
            {
                public void actionPerformed(ActionEvent e)
                {
                    try
                    {
                        String testString =
                            _startThrustTextField.getText();
                        initialThrust = factor *
                            Double.valueOf(testString).doubleValue();
                        thrustErrorMessage.setText ("");
                        initThrustSet = true;
                        _endThrustTextField.requestFocus();
                    }
                    catch (NumberFormatException error)
                    {
                        thrustErrorMessage.setText ("Enter a
                            number");
                        initThrustSet = false;
                    }
                }
            });
            thrustPanel.add(_startThrustTextField);
        }
    }
}

```

```
    }
    else
    {
        trajectory.setSegment (_segmentArrayList);
        _segmentsSet = true;
    }
}
} // SetUp class
```
